

Analysis of Thermal and Reaction Times for Hydrogen Reduction of Lunar Regolith

U. Hegde and R. Balasubramaniam National Center for Space Exploration Research, Glenn Research Center, Cleveland, Ohio

S. Gokoglu Glenn Research Center, Cleveland, Ohio

NASA STI Program . . . in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) program plays a key part in helping NASA maintain this important role.

The NASA STI Program operates under the auspices of the Agency Chief Information Officer. It collects, organizes, provides for archiving, and disseminates NASA's STI. The NASA STI program provides access to the NASA Aeronautics and Space Database and its public interface, the NASA Technical Reports Server, thus providing one of the largest collections of aeronautical and space science STI in the world. Results are published in both non-NASA channels and by NASA in the NASA STI Report Series, which includes the following report types:

- TECHNICAL PUBLICATION. Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- TECHNICAL MEMORANDUM. Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- CONTRACTOR REPORT. Scientific and technical findings by NASA-sponsored contractors and grantees.
- CONFERENCE PUBLICATION. Collected

papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.

- SPECIAL PUBLICATION. Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- TECHNICAL TRANSLATION. Englishlanguage translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services also include creating custom thesauri, building customized databases, organizing and publishing research results.

For more information about the NASA STI program, see the following:

- Access the NASA STI program home page at http://www.sti.nasa.gov
- E-mail your question via the Internet to *help@ sti.nasa.gov*
- Fax your question to the NASA STI Help Desk at 301–621–0134
- Telephone the NASA STI Help Desk at 301–621–0390
- Write to: NASA Center for AeroSpace Information (CASI) 7115 Standard Drive Hanover, MD 21076–1320



Analysis of Thermal and Reaction Times for Hydrogen Reduction of Lunar Regolith

U. Hegde and R. Balasubramaniam National Center for Space Exploration Research, Glenn Research Center, Cleveland, Ohio

S. Gokoglu Glenn Research Center, Cleveland, Ohio

Prepared for the Space Technology and Applications International Forum (STAIF–2008) sponsored by the Institute for Space and Nuclear Power Studies at the University of New Mexico Albuquerque, New Mexico, February 10–14, 2008

National Aeronautics and Space Administration

Glenn Research Center Cleveland, Ohio 44135

Acknowledgments

 U. Hegde and R. Balasubramaniam were supported under NASA Cooperative Agreement NCC3–975 with the National Center for Space Exploration Research. The authors greatly appreciate the encouragement and insights offered by K. Sacksteder and D. Linne of the NASA Glenn Research Center, and useful discussions with E. Santiago-Maldonado of the NASA Kennedy Space Center during the course of this work.

Level of Review: This material has been technically reviewed by technical management.

Available from

NASA Center for Aerospace Information 7115 Standard Drive Hanover, MD 21076–1320 National Technical Information Service 5285 Port Royal Road Springfield, VA 22161

Available electronically at http://gltrs.grc.nasa.gov

Analysis of Thermal and Reaction Times for Hydrogen Reduction of Lunar Regolith

U. Hegde and R. Balasubramaniam National Center for Space Exploration Research Glenn Research Center Cleveland, Ohio 44135

S. Gokoglu National Aeronautics and Space Administration Glenn Research Center Cleveland, Ohio 44135

Abstract

System analysis of oxygen production by hydrogen reduction of lunar regolith has shown the importance of the relative time scales for regolith heating and chemical reaction to overall performance. These values determine the sizing and power requirements of the system and also impact the number and operational phasing of reaction chambers. In this paper, a Nusselt number correlation analysis is performed to determine the heat transfer rates and regolith heat up times in a fluidized bed reactor heated by a central heating element (e.g., a resistively heated rod, or a solar concentrator heat pipe). A coupled chemical and transport model has also been developed for the chemical reduction of regolith by a continuous flow of hydrogen. The regolith conversion occurs on the surfaces of and within the regolith particles. Several important quantities are identified as a result of the above analyses. Reactor scale parameters include the void fraction (i.e., the fraction of the reactor volume not occupied by the regolith particles) and the residence time of hydrogen in the reactor. Particle scale quantities include the particle Reynolds number, the Archimedes number, and the time needed for hydrogen to diffuse into the pores of the regolith particles. The analysis is used to determine the heat up and reaction times and its application to NASA's oxygen production system modeling tool is noted.

Nomenclature

| A | reactor cross section area (m ²) |
|---------------|---|
| $A_{\rm eff}$ | effective surface area of heater (m ²) |
| Ar | Archimedes number |
| С | molar concentration of hydrogen in particle (moles/m ³) |
| C_{S} | molar concentration of hydrogen at reaction surface (moles/m ³) |
| C_x | molar concentration of hydrogen at location x in reactor (moles/m ³) |
| C_0 | molar concentration of hydrogen at upstream end of regolith bed (moles/m ³) |
| $C_{\rm eff}$ | effective specific heat (J/kg/K) |
| C_{pg} | specific heat of gas (J/kg/K) |
| C_{ps} | specific heat of regolith (J/kg/K) |
| D | effective gas diffusion coefficient (m ² /s) |
| Ε | energy required to heat bed |
| h | heat transfer coefficient |
| g k | gravitational acceleration (m/s ²) |
| k | equilibrium constant |
| L | length of regolith bed (m) |
| $M_{ m eff}$ | effective mass (kg) |

| 1.6 | |
|-----------------------------|---|
| $M_{ m total}$ | total molar water yield (mol) |
| N | number of regolith particles in the bed |
| Nu | Nusselt number |
| Pr | Prandtl number |
| r_p | regolith particle radius (m) |
| Re | Reynolds number |
| t | time (s) |
| t_c | chemical time (s) |
| t_D | intra-particle diffusion time (s) |
| <i>t</i> _{DLimit} | chemical time in diffusion limit (s) |
| t_R | reactor residence time (s) |
| <i>t_{ResLimit}</i> | chemical time in equilibrium limit (s) |
| Т | temperature (K) |
| T_h | heater temperature (K) |
| T_1, T_2 | initial and final temperature of bed (K) |
| и | velocity through reactor (m/s) |
| $V_{\rm bed}$ | volume of regolith bed (m^3) |
| x | axial coordinate (m) |
| α | particle conversion |
| α_0 | particle conversion at $x = 0$ |
| β | regolith thermal diffusivity (m ² /s) |
| δ | thermal conduction length scale (m) |
| 3 | void fraction in regolith bed |
| ρ_a | molar concentration of ilmenite in regolith (mol/m ³) |
| $ ho_g$ | gas density (kg/m ³) |
| ρ_s | regolith density (kg/m ³) |
| μ | gas viscosity (Ns/m ²) |
| λ | gas thermal conductivity |
|)T/ | malar autflaur of water (mal/a) |

 Ψ molar outflow of water (mol/s)

Introduction

Hydrogen reduction of lunar regolith has been proposed as a viable technology for oxygen production on the Moon. (Taylor and Carrier 1992, McKay and Allen 1996, Rice, Hermes and Musbah 1997). The reduction processes involve solid-gas reactions. Such systems are well developed for earth applications but mainly for the purpose of metal extraction as opposed to water vapor and oxygen production for lunar applications. In addition, in the lunar context, the reactor must interface with the other sub-system processes such as upstream regolith extraction and beneficiation, and downstream electrolysis and phase separation in a way that establishes the most favorable balance between efficiency, robustness, and equivalent system mass. For example, the reduction of ilmenite may be represented as:

$$FeO.TiO_2 + H_2 \Leftrightarrow Fe + TiO_2 + H_2O$$
(1)

It is carried out in the reactor sub-system. The formed water may be electrolyzed in the electrolyzer subsystem to produce oxygen and hydrogen.

From upmass and cost considerations, it is essential to operate these systems on the lunar surface in as nearly a closed-loop manner as possible for the hydrogen. It is in this context that knowledge of processing times in the various sub-systems becomes important. For example, a mismatch between the flux of water exiting the reactor and the amount of water that may be processed by the electrolyzer must be mitigated by design such as sizing the appropriate buffer tank. This paper investigates the factors that influence the overall processing time in the reactor sub-system.

The temperature of the reaction is one of the important parameters of the process. Ilmenite is just one constituent of mare basalt and it cannot be assumed that regolith beneficiation (e.g., in this case, removal of the nonilmenite components) will be carried out prior to the reactor processes. The sintering and melting temperature of the unbeneficiated regolith is noted to be in the range of 1370 to 1670 K (Chambers 1995, McKay and Allen 1996). However, hydrogen reduction has been found to be effective also at lower temperatures, e.g., in the range of 970 to 1270 K (Yoshida et al., 2000). Therefore, while three-phase (gas-liquid-solid) reactors for hydrogen reduction may be applicable at post-melt temperatures, these are not considered here and attention is restricted to gas-solid interactions. In order to provide a basis for discussion, it is assumed that the chemical reduction of the regolith occurs in a gas-solid fluidized bed reactor.

The conversion process in the reactor consists of the following overall steps (1) transfer of regolith into the reactor, (2) fluidization and heating of the regolith, (3) reaction of hydrogen with the ilmenite portion of the regolith, (4) outflow of left-over hydrogen and produced water vapor, and (5) removal of the spent regolith from the reactor. In this paper, attention is focused on steps (2) to (4).

The reported values for the thermal diffusivity, β , of lunar regolith are very low (Heiken, Vaniman and French 1991). In the absence of a gas flow, the time to heat the regolith bed over a distance δ is proportional the ratio δ^2/β , which results in unacceptably large values. On the other hand, hydrogen has a relatively high thermal conductivity and helps to stir the bed so that having hydrogen flow through the bed decreases the heating time appreciably. Thus, it is advantageous to fluidize the bed during the regolith heating step. There are also advantages related to reducing regolith particle agglomeration and sintering which can adversely impact reaction times and water production rates.

The objectives of this paper are (1) to develop analytical models for regolith heating and for chemical reduction by hydrogen in order to identify parameters influencing the heating and reaction times, (2) to identify the parameters that govern these times, and (3) to obtain relationships for the reaction time in the two limits of diffusion (i.e., mass transfer) control and chemical equilibrium control. As noted earlier, these times are important from the overall oxygen production system perspective. The thermal analysis is based upon a Nusselt number correlation. The reaction model considered is based upon a "shrinking core" diffusion-reaction model for the gas-solid reduction process (Levenspiel 1998) which is coupled with the inflow/outflow of gases. The methodology has been discussed in our earlier work (Hegde, Balasubramaniam and Gokoglu 2007). It appears consistent with results of regolith particulate analysis conducted on Apollo-returned samples (McKay and Allen 1996) which suggest that the ilmenite particles are not completely solid but have some porosity that would enable gases to diffuse into the particles, i.e., the FeO portion can be essentially completely converted leaving behind the reduced iron.

In the next section, the regolith heating is considered first and a model for the heating time and required power is developed. Pertinent results from the shrinking core analysis are also presented. Subsequently, in the Results section, factors impacting these processes are evaluated and discussed and applications of the model to NASA's oxygen production system modeling tool are noted.

Model Development

In the first sub-section below, the time taken to heat the regolith is calculated using a Nusselt number correlation for fluidized bed heat transfer. This is followed by a summary of the particle conversion analysis and the coupling between the particle conversion and the bulk reactor flow.

Heat Transfer Analysis

It is assumed that the regolith is heated by a cylindrical rod located centrally in the cylindrical reactor. The surface temperature of the heater rod can be controlled e.g., by adjusting the current through it as in an electrical heater or by controlling the energy flux as in a solar concentrator heat pipe. The different

pathways for heat transfer in the bed are gas convection and conduction, particle motions, and possibly, radiative transfer.

Fluidized bed heat transfer correlations developed in the literature under convective-conductive conditions suggest that the controlling step is the heat transfer from the heater surface to the gas film next to it. Direct transfer of heat to the bed particles adjoining the heater is not efficient because the contact area between particles and the heater surface is small. If the bed is well stirred by fluidization, its temperature is nearly uniform. In such a case, a Nusselt number may be defined which relates the heat transfer to the bed through the gas film in contact with the heater, i.e., the regolith bed is taken to be a uniform entity. The reactor wall is not usually insulated from the bed, so in general, there will also be heat transfer from the bed to the reactor wall. If the reactor wall is not too thick and its thermal conductivity is high, the wall may be assumed to be at a uniform temperature. Furthermore, if the reactor's thermal mass (i.e., the product of its mass and specific heat) is small compared to that of the bed, it is reasonable to assume that the temperature of the reactor is the same as that of the bed.

The heating rate with the above scenario is given by:

$$M_{\rm eff}C_{\rm eff}\frac{dT}{dt} = hA_{\rm eff}(T_h - T), \qquad (2)$$

where the thin film heat transfer coefficient, h, is related to the Nusselt number, Nu, by:

$$Nu = 2hr_p / \lambda . ag{3}$$

In Equation (2), $M_{\text{eff}}C_{\text{eff}}$ is the effective value of the product of the mass and specific heat and is a weighted average of the values for the gas, regolith, and reactor. A_{eff} is the effective or active surface area of the heater rod in contact with the bed.

The solution of Equation (2) for the instantaneous bed temperature *T* is given by:

$$T = T_h - (T_h - T_1) \exp\left(-hA_{\rm eff}t / (M_{\rm eff}C_{\rm eff})\right).$$
(4)

Conversely, the time to raise the bed temperature from T_1 , to a temperature T_2 is given by:

$$t_h = -\frac{M_{\rm eff}C_{\rm eff}}{hA_{\rm eff}}\ln\left(\frac{T_h - T_2}{T_h - T_1}\right).$$
(5)

Various fluidized bed heat transfer Nusselt number correlations are available in the literature. For example, the following one (Borodulya et al. 1991) appears to describe the situation adequately:

Nu = 0.74Ar^{0.1}
$$\left(\frac{\rho_s}{\rho_g}\right)^{0.14} \left(\frac{C_{ps}}{C_{pg}}\right)^{0.24} (1-\epsilon)^{2.3} + 0.46 \operatorname{Re} \operatorname{Pr} \frac{(1-\epsilon)^{2.3}}{\epsilon}$$
. (6)

The Archimedes number, Ar, the Reynolds number, Re, and the Prandtl number, Pr, are defined as follows:

$$Ar = 8gr_p^3 \rho_g (\rho_s - \rho_g)/\mu^2, \qquad (7)$$

$$\operatorname{Re} = 2ur_p \rho_g / \mu \quad , \tag{8}$$

NASA/TM-2009-215623

$$\Pr = C_{pg} \mu / \lambda_g \,. \tag{9}$$

The energy, *E*, required to heat the bed and reactor from T_1 to T_2 , in the absence of other losses (e.g., radiative), is given by:

$$E = M_{\rm eff} C_{\rm eff} (T_2 - T_1) \,. \tag{10}$$

An average power required is obtained by dividing *E* by the time, T_h . Note that the instantaneous power, $M_{\text{eff}}C_{\text{eff}}DT/dt$, can be obtained by substituting the solution for *T* (Eq. (4)) into Equation (2).

Since radiative losses vary as the fourth power of temperature, they can become important as the reactor is heated to higher temperatures. It is expected that there would be insulation around the reactor that would reduce this loss. This loss can be accounted for by including it in the energy required.

Chemical Conversion Analysis

The chemical conversion analysis couples the "shrinking core" model with the inflow and outflow of the gases and has been discussed in earlier work (Hegde, Balasubramaniam and Gokoglu, 2007). Here, some results are presented to highlight the findings from the model and to show how the characteristic times of the process are obtained.

Defining the conversion α by:

$$\alpha = 1 - \frac{r_s^3}{r_p^3},\tag{11}$$

the shrinking core analysis of a single particle yields a relationship between α and the associated conversion time, *t*:

$$3[1-\alpha]^{2/3} - 2[1-\alpha] - 1 = -\frac{6D[c_x - c_s]t}{\rho_a r_p^2} \quad .$$
(12)

Further, assuming equilibrium conditions at the reaction surface, the above equation reduces to:

$$3[1-\alpha]^{2/3} - 2[1-\alpha] - 1 = -\frac{6Dkc_0t}{(1+k)\rho_a r_p^2}.$$
(13)

Note that an expression for the equilibrium constant, k, as a function of only the reaction temperature has been provided (William and Erstfeld, 1979):

$$k = 10^{\left(-2126.1/T + 0.6439\right)}.$$
(14)

The time for complete conversion is given by setting α to unity.

Coupling Particle Analysis With the Reactor Bulk Flow

Assuming that the gas velocity, concentration of hydrogen and the associated conversion α are functions of axial distance along the reactor, and the inlet hydrogen concentration is constant in time, the following equation has been obtained:

$$-4\ln\left[\frac{1-(1-\alpha)^{1/3}}{1-(1-\alpha_0)^{1/3}}\right] + \ln\left[\frac{2(1-\alpha)^{1/3}+1}{2(1-\alpha_0)^{1/3}+1}\right] = 3(1-\varepsilon)\frac{D}{r_p^2 u}x,$$
(15)

where α_0 is the value of α at x = 0 and is given by Equation (13).

In certain cases, closed form solutions may be obtained for the concentration of hydrogen and extent of conversion in the reactor. An example is when the regolith and gases are sufficiently stirred or wellmixed so that gradients in conversion and gas concentration are smeared out. In this well-stirred limit, the concentration of hydrogen exiting the reactor is also the average concentration in the reactor and is given by:

$$c(L) = \frac{c_0}{1+k} \left\{ 1 + \frac{k}{\left[1+3(1-\varepsilon)\left(\frac{D}{r_p^2}\right) \frac{[1-\alpha]^{1/3}}{[1-(1-\alpha)^{1/3}]^u} \right]^2} \right\}.$$
 (16)

Results

Regolith batch sizing will depend upon the oxygen production required and the processing time per batch. In general, the processing time per batch is the sum of the heating and chemical conversion times. This section considers factors influencing these times based upon the analysis presented earlier.

Heating Time

Equation (5) indicates that the heating time is related directly to the Nusselt number and batch size (i.e., regolith mass) and inversely related to the active surface area of the heater. The Nusselt number depends upon the Reynolds number of the gas flow relative to the regolith particles, the Archimedes number, and the Prandtl number of the fluidizing gas. The Prandtl number is practically constant with temperature. The Reynolds number is related to the fluidizing velocity and particle size. The particle size is fixed by the regolith characteristics and there may be some flexibility in this regard depending upon the preprocessing facilities available. The fluidizing velocity is also related to the particle size. The Archimedes number is mainly dependent upon the regolith density (which is much greater than the gas density). The most flexible way of changing the heating time seems to be by changing the regolith batch size, active heater area, and/or power input to the heater.

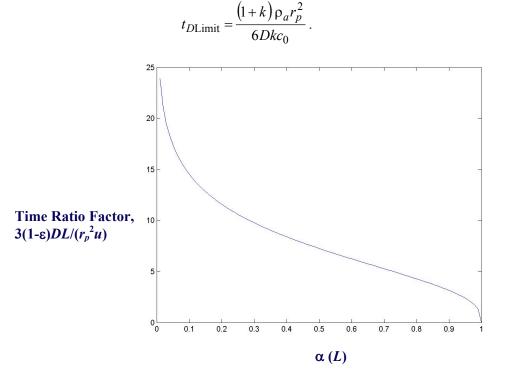
The active heater surface area may be increased by increasing the heater diameter. For a fixed reactor diameter, this also increases the height, L, of the regolith bed which is also conducive to heat transfer. However, this may have implications for the reaction time as will be discussed later. The regolith batch

size may be reduced to decrease the heating time but this may be restricted from overall oxygen production considerations.

The power input to the heater is related to the heater temperature T_h . From Equation (5), it is clear that increasing the heater temperature will decrease the time to heat the regolith to a given temperature T_2 . However, there are limitations to raising T_h substantially over T_2 e.g., the presence of hot spots or regions in the regolith bed which are at higher than desired temperature.

Chemical Time

The shrinking core analysis provides an expression for the conversion time of a single particle. In many instances it is assumed that the conversion time of the regolith bed is the same as the single particle conversion time. Equation (15) provides a means for determining the conditions under which this holds. Choosing x = L (i.e., the height of the regolith bed), it is clear that the right hand side of Equation (15) is related to the ratio of the gas residence time in the bed, L/u, to its diffusion time in a particle, r_p^2/D . This time factor ratio plays an important role in determining the extent of conversion in the bed. For example, the upstream end (entrance region) of the bed is always exposed to fresh hydrogen and would be the first to get converted. The time ratio factor gives a measure of the conversion in the remaining portions of the bed. Figure 1 plots the conversion, $\alpha(L)$, at the downstream end (exit region) as a function of the time ratio factor for the case where α_0 is equal to unity, i.e., when the particles at the entrance are already completely converted. As the time ratio factor becomes small (i.e., the intra-particle diffusion time is much greater than the gas residence time in the regolith bed) it can be seen from Figure 1 that the conversion, $\alpha(L)$, also approaches unity, i.e., the particles at the downstream end of the bed (and more generally, at any location in the bed) are also consumed fully. This indicates that the time rate of chemical conversion for all particles in the bed is given by Equation (13). The corresponding time for complete conversion may be termed the chemical time in the diffusion controlled limit. Setting α to one in Equation (13) yields:



(17)

Figure 1.—Dependence of the Regolith Conversion at the Downstream End of the Reactor Bed Upon the Time Ratio Factor $3(1-\varepsilon) DL/r_{\rho}^{2}u$.

On the other hand, as the residence time becomes much larger compared to the diffusion time, i.e., the time ratio factor becomes large, it can be seen from Figure 1 that the conversion occurs first primarily near the upstream end of the bed and not much conversion occurs downstream since $\alpha(L)$ is small for this condition. This is referred to as the chemical equilibrium limit, i.e., the residence time in the bed is long enough so that the water concentration reaches its chemical equilibrium value in the gas phase. At this point no further conversion takes place. In this limiting case, the chemical time may be estimated by assuming equilibrium concentration of water in the outflow. Then, the molar outflow of water is given by:

$$\Psi = \frac{k}{(1+k)} u A c_0 , \qquad (18)$$

and the total molar water yield is:

$$M_{\text{total}} = \psi t_{\text{ResLimit}} . \tag{19}$$

Equating this to the available molar water yield in the regolith provides the chemical time in the residence time limit or equilibrium limit:

$$t_{\text{ResLimit}} = \rho_a (1 - \varepsilon) V_{\text{bed}} / \psi .$$
⁽²⁰⁾

Since the conversion time in the diffusion limit depends inherently upon the particle characteristics such as its size and the intra-particle gas diffusivity, it is clear that this is a lower limit to the conversion time. The equilibrium limiting conversion time is variable however, depending upon the flow velocity and bed height. Varying these quantities is a way to slow down the conversion rate. This may be useful from an overall system design perspective.

Application to System Design

The analysis of heating and chemical times presented here may be applied to sizing and design of the reactor assembly. It has been noted in earlier studies that the regolith reduction rate increases directly with the temperature, (McKay and Allen 1996) presumably due to increase in the effective diffusion coefficient (Hegde, Balasubramaniam and Gokoglu 2007). This implies that most of the conversion occurs at the operating temperature. In other words, the heating and chemical reductions are essentially serial operations for a given batch of regolith. Thus, a system comprising two reactors may be used for increased throughput. Ignoring for current purposes, the regolith loading and unloading times into the reactor chamber, it is clear that designing for equal heating and reaction times will ensure continuous production of oxygen, i.e., as one batch of regolith is being heated the other batch is being chemically reduced.

The equations provided here provide a framework for achieving the proper design parameters. The diffusion limited chemical time (Eq. (17)) provides the minimum time for conversion. Equations (5) and (10) in conjunction provide the power and heater rod characteristics (e.g., effective area) for a given heating time, e.g., the minimum reaction time. If the power available is less than the heating power required, the heating time has to be increased. In such a case, the reaction time needs to be modified e.g., by driving the reactor dimensions toward the residence limited conversion time (Eq. (20)). Clearly, this is an iterative process. Currently, NASA is developing an overall system model incorporating these ideas (Santiago-Maldonado 2007).

Conclusions

Analytical models for regolith heating and for chemical reduction by hydrogen have been developed to identify parameters influencing the heating and reaction times. These times are important from the overall oxygen production system perspective. The regolith heating model is based upon a Nusselt number correlation for fluidized bed heat transfer. While particle size and parameters such as the particle Reynolds and Archimedes numbers play a role in the heating time determination, it appears that the most feasible way of controlling this time is by adjusting the active heater surface area and mass of the regolith per batch. The chemical reduction model results from coupling a shrinking core type of analysis with the overall regolith bed flow characteristics. It is shown that the chemical time in the diffusion limit is also the lower limit for the conversion time. However, by adjusting the residence time of the gases in the regolith bed, the conversion time can be increased.

References

- Borodulya, V.A., Teplitsky, Y.S., Markevich, I.I., Hassan, A.F., and Yeryomenko, T.P., "Heat Transfer Between a Surface and a Fluidized Bed: Consideration of Pressure and Temperature Effects," *International Journal of Heat and Mass Transfer*, 34, 47–53, (1991).
- Chambers, J.G., "Quantitative Mineralogical Characterization of Lunar High-Ti Mare Basalts and Soils for Oxygen Production," *Journal of Geophysical Research*, 100, 14391–14401, (1995).
- Hegde, U., Balasubramaniam, R., and Gokoglu, S., *Development of a Reactor Model for Chemical Conversion of Lunar Regolith*, Space Technologies and Applications International Forum, Albuquerque, NM, 2007.
- Heiken, G.H. Vaniman, D.T., and French, B.M., *Lunar Sourcebook*, Cambridge University Press, New York, NY, 1991, pp. 285–300.
- Levenspiel, O., *Chemical Reaction Engineering*, John Wiley & Sons, New York, NY, 1998, pp. 424–498.
- McKay, D.S., and Allen, C.C., Hydrogen Reduction of Lunar Materials for Oxygen Extraction on the Moon, AIAA Paper No. 96–0488, American Institute of Aeronautics and Astronautics, Washington, DC, 1996.
- Rice, E.E., Hermes, P.A., and Musbah, O.A., *Carbon Based Reduction of Lunar Oxides for Oxygen Production*, AIAA Paper No. 97–0890, American Institute of Aeronautics and Astronautics, Washington, DC, 1997.
- Santiago-Maldonado, E., ISRU SystemModel Tool: From Excavation to Oxygen Production, Space Resources Roundtable Inc., Golden, CO, 2007.
- Taylor, L.A., and Carrier, W.D. III, "Production of Oxygen on the Moon: Which Processes are Best and Why," *AIAA Journal*, 30, (12), 2858–2863, (1992).
- Williams, R.J., and Erstfeld, T.E., *High Temperature Electrolyte Recovery of Oxygen From Gaseous Effluents From the Carbochlorination of Lunar Anthracite and the Hydrogenation of Ilmenite: A Theoretical Study*, NASA TM 58214, NASA, Washington, DC, 1979.
- Yoshida, H., Watanabe, T., Kanamon, H., Yoshida, T., Ogiwara, S., and Eguchi, K., *Experimental Study* on Water Production by Hydrogen Reduction of Lunar Soil Simulant in a Fixed-Bed Reactor, Space Resources Roundtable Inc., Golden, CO, 2000.

| In begin production but collection of information, its estimated to average 1 how zerr response, including the time for everymen and the collection of information, infor | educing this 2-4302. |
|---|---|
| 1. REPORT DATE (DD-MM-YYYY) 2. REPORT TYPE 3. DATES COVERED (From - To) 01-04-2009 Technical Memorandum 5a. CONTRACT NUMBER Analysis of Thermal and Reaction Times for Hydrogen Reduction of Lunar Regolith 5a. CONTRACT NUMBER 5b. GRANT NUMBER 5b. GRANT NUMBER 6. AUTHOR(S) 5d. PROGRAM ELEMENT NUMBER 1. Hegde, U.; Balasubramaniam, R.; Gokoglu, S. 5d. PROJECT NUMBER 5c. TASK NUMBER 5d. NORK UNIT NUMBER 9. AUTHOR(S) 5d. NORK UNIT NUMBER 9. SPONSORING ORGANIZATION NAME(S) AND ADDRESS(ES) 8. PERFORMING ORGANIZATION REPORT NUMBER 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 8. PERFORMING ORGANIZATION REPORT NUMBER 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 2. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration NASA 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) NASA National Aeronautics and Space Administration NASA National Aeronautics and Space Administration NASA | |
| 4. TITLE AND SUBTITLE Sa. CONTRACT NUMBER Analysis of Thermal and Reaction Times for Hydrogen Reduction of Lunar Regolith 5a. CONTRACT NUMBER 5b. GRANT NUMBER 5b. GRANT NUMBER 5c. PROGRAM ELEMENT NUMBER 5c. PROGRAM ELEMENT NUMBER 6. AUTHOR(S) 5d. PROJECT NUMBER 9. AUTHOR(S) 5d. PROJECT NUMBER 9. TASK NUMBER 5f. WORK UNIT NUMBER 9. SPONSORING ORGANIZATION NAME(S) AND ADDRESS(ES) 8. PERFORMING ORGANIZATION RESORT NUMBER 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 8. PERFORMING ORGANIZATION REPORT NUMBER 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSORING/MONITORS ACRONYM(S) National Acronautics and Space Administration Washington, DC 20546-0001 10. SPONSORING/MONITORS ACRONYM(S) 12. DISTRIBUTION/AVAILABILITY STATEMENT | |
| 6. AUTHOR(S) 5c. PROGRAM ELEMENT NUMBER Hegde, U.; Balasubramaniam, R.; Gokoglu, S. 5d. PROJECT NUMBER 5e. TASK NUMBER 5e. TASK NUMBER 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) 5f. WORK UNIT NUMBER National Aeronautics and Space Administration 8. PERFORMING ORGANIZATION RAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 8. PERFORMING ORGANIZATION RAME(S) AND ADDRESS(ES) 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration ACRONTM(S) Washington, DC 20546-0001 10. SPONSORING/MONITORING REPORT NUMBER Vundassified-Unlimited NASA 11. SPONSORING/MONITORING agency name (S) AND ADDRESS(ES) NASA 12. DISTRIBUTION/AVAILABILITY STATEMENT NASA Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| 6. AUTHOR(S) Hegde, U.; Balasubramaniam, R.; Gokoglu, S. 5d. PROJECT NUMBER 5e. TASK NUMBER 5e. TASK NUMBER 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191 8. PERFORMING ORGANIZATION REPORT NUMBER E-16705-1 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field 10. SPONSORING/MONITORS 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, DC 20546-0001 10. SPONSORING/MONITORS ACRONYM(S) NASA 11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-215623 11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-215623 12. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| Hegde, U.; Balasubramaniam, R.; Gokoglu, S. 5e. TASK NUMBER 5e. TASK NUMBER 5f. WORK UNIT NUMBER WBS 387498.04.01.04.03 5f. WORK UNIT NUMBER National Aeronautics and Space Administration 8. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 8. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 8. PERFORT NUMBER Cleveland, Ohio 44135-3191 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration 10. SPONSORING/MONITORING MONITORING REPORT NUMBER National Aeronautics and Space Administration NASA Washington, DC 20546-0001 11. SPONSORING/MONITORING MONITORING MONITOR | |
| 57. WORK UNIT NUMBER WBS 387498.04.01.04.03 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191 8. PERFORMING ORGANIZATION REPORT NUMBER E-16705-1 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, DC 20546-0001 10. SPONSORING/MONITORS ACRONYM(S) NASA 11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-215623 11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-215623 22. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| WBS 387498.04.01.04.037. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-31918. PERFORMING ORGANIZATION REPORT NUMBER E-16705-19. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, DC 20546-000110. SPONSORING/MONITORS ACRONYM(S) NASA11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-21562311. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-21562312. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| National Aeronautics and Space Administration REPORT NUMBER John H. Glenn Research Center at Lewis Field E-16705-1 Cleveland, Ohio 44135-3191 10. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration National Aeronautics and Space Administration Washington, DC 20546-0001 10. SPONSORING/MONITORING REPORT NUMBER REPORT NUMBER NASA 11. SPONSORING/MONITORING Report NUMBER National Aeronautics and Space Administration Washington, DC 20546-0001 11. SPONSORING/MONITORING Report NUMBER NASA 11. SPONSORING/MONITORING Report NUMBER NASA/TM-2009-215623 12. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| National Aeronautics and Space Administration ACRONYM(S) Washington, DC 20546-0001 NASA 11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2009-215623 NASA/TM-2009-215623 12. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 Subject Category: 30 | 1 |
| REPORT NUMBER NASA/TM-2009-215623 12. DISTRIBUTION/AVAILABILITY STATEMENT Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| Unclassified-Unlimited Subject Category: 31 Available electronically at http://gltrs.grc.nasa.gov This publication is available from the NASA Center for AeroSpace Information, 301-621-0390 | |
| 13. SUPPLEMENTARY NOTES | |
| | |
| | |
| 14. ABSTRACT System analysis of oxygen production by hydrogen reduction of lunar regolith has shown the importance of the relative time scales fregolith heating and chemical reaction to overall performance. These values determine the sizing and power requirements of the syst also impact the number and operational phasing of reaction chambers. In this paper, a Nusselt number correlation analysis is perform determine the heat transfer rates and regolith heat up times in a fluidized bed reactor heated by a central heating element (e.g., a resist heated rod, or a solar concentrator heat pipe). A coupled chemical and transport model has also been developed for the chemical reduced regolith by a continuous flow of hydrogen. The regolith conversion occurs on the surfaces of and within the regolith particles. Several important quantities are identified as a result of the above analyses. Reactor scale parameters include the void fraction (i.e., the fraction the particle Reynolds number, the Archimedes number, and the time needed for hydrogen to diffuse into the pores of the regolith particles number, and the time needed for hydrogen to diffuse into the pores of the regolith particle. 15. SUBJECT TERMS Hydrogen reduction; ISRU; Lunar regolith; Ilmenite; Shrinking core | em and hed to trively action of al on of include ticles. |
| 16. SECURITY CLASSIFICATION OF: 17. LIMITATION OF 18. NUMBER 19a. NAME OF RESPONSIBLE PERSO |)N |
| a. REPORT b. ABSTRACT c. THIS OF STI Help Desk (email:help@sti.nas U PAGE UU 15 19b. TELEPHONE NUMBER (include and 301-621-0390 | |

Г