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PROGRESS REPORT

Air Conditioning in Underground Structures

November 1, 1952 to April 30, 1953.

by

B. A. Peavy
Heating and Air Conditioning Section
Building Technology Division.



U. S. DEPARTMENT OF COMMERCE
NATIONAL BUREAU OF STANDARDS

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PROGRESS REPORT

Air Conditioning in Underground Structures

I. INTRODUCTION

During the period from November 1, 1952 to April 31, 1953, tests have been continued in the underground test chamber at Mount Weather, mathematical approaches to the theory of heat transfer to a rock mass bounding an underground space have been studied and are compiled and presented in this report, data from previous tests were analyzed, revisions were made in the Engineering Manual, Part XVI, Chapter 3, and preparations were made for future testing at Mount Weather, Va. and Fort Ritchie, Maryland.

II. TESTS PERFORMED IN UNDERGROUND TEST CHAMBER

- Test Condition 8 - Determination of heat and moisture load without ventilation or simulated occupancy from November 10 to November 26 while maintaining constant conditions of 75° DB and 50% R.H. with the air conditioning system.
- Test Condition 9 - Conditions were the same as test condition 8 except condenser water reheat was used to reduce the electric reheat November 26 to December 5.
- Test Condition 10 - Steady state heat at 75°F air temperature from December 5 to January 5, 1953. No dehumidification.
- Test Condition 11 - Steady state heating at 75°F air temperature with ventilation air from January 5 to February 11.
- Test Condition 12 - Test condition 10 repeat from February 11 to March 30.
- Test Condition 13 - Test condition 11 repeat now in progress.

1. Description of the equipment

1.1. Description

During the period from November 1, 1955 to April 11, 1957, tests have been conducted in the laboratory for the purpose of determining the characteristics of the equipment under test. The results of these tests are given in the following tables. The tests were conducted in the laboratory of the Department of Defense, and the results are given in the following tables. The tests were conducted in the laboratory of the Department of Defense, and the results are given in the following tables.

1.2. Test results

Test results are given in the following tables. The tests were conducted in the laboratory of the Department of Defense, and the results are given in the following tables. The tests were conducted in the laboratory of the Department of Defense, and the results are given in the following tables.

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III. MATHEMATICAL APPROACHES TO HEAT TRANSFER TO UNDERGROUND SPACES

A review of the mathematical approaches to the transfer of heat to the rock surrounding an underground chamber shows that mathematical treatment of the elementary shapes such as the plane surface, circular cylinder and sphere bounded by a medium approaching infinite thickness is found in Carslaw and Jaeger, "Conduction of Heat in Solids". While for the most part the actual equations involving heat transfer are complicated, they may be reduced by numerical integration or approximation and plotted in graphical or tabular form for design application.

Configuration of actual underground installations may be approximated by one or more of the elementary shapes. Considering an underground room to be an assembly of plane surfaces (floor, ceiling and walls) neglects the heat flow into the corners and edges, whereas likening it to a cylinder (lateral surface) neglects the heat flow into ends of the cylinder. A sphere would rarely be an approximation for an actual installation.

To the above named shapes two different boundary conditions can be applied; namely, the constant heat flux case and the constant surface temperature case.

I. Heat transferred to a solid bounding an elementary shape with a constant heat flux and the initial temperature of the solid equal to zero on an arbitrary datum plane.

A. Linear heat flow to a medium of semi-infinite depth from a plane surface.

1. Temperature at depth x

$$\theta = \frac{Q}{K} \sqrt{\alpha t} \left(2 \operatorname{ierfc} \frac{x}{2 \sqrt{\alpha t}} \right)$$

Values of '2 ierfc' of the argument are shown in tabular form in Table 2.

III. EXPERIMENTAL INVESTIGATION OF THE EFFECTS OF CHLORINE ON THE

A series of the following experiments was
conducted to determine the effect of chlorine
on the growth of the plant. The plants were
grown in a soil containing a certain amount
of chlorine. The plants were divided into
two groups. One group was grown in a soil
containing a certain amount of chlorine, and
the other group was grown in a soil
containing no chlorine. The plants were
grown for a certain period of time, and
the results were compared.

The results of the experiments were as follows:
The plants grown in a soil containing a
certain amount of chlorine grew to a
certain height, and the leaves were of a
certain color. The plants grown in a soil
containing no chlorine grew to a certain
height, and the leaves were of a certain
color. The results of the experiments
showed that the plants grown in a soil
containing a certain amount of chlorine
grew to a certain height, and the leaves
were of a certain color.

To the above results, the following
conclusions were drawn: The plants grown
in a soil containing a certain amount of
chlorine grew to a certain height, and the
leaves were of a certain color.

I. The results of the experiments
showed that the plants grown in a soil
containing a certain amount of chlorine
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were of a certain color.

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showed that the plants grown in a soil
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showed that the plants grown in a soil
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were of a certain color.

$$\frac{1}{\sqrt{x}} = \frac{1}{\sqrt{x}} \cdot \frac{\sqrt{x}}{\sqrt{x}} = \frac{\sqrt{x}}{x}$$

The results of the experiments
showed that the plants grown in a soil
containing a certain amount of chlorine
grew to a certain height, and the leaves
were of a certain color.

2. Temperature at $x = 0$.

$$\theta = \frac{2Q}{k} \sqrt{\frac{\alpha t}{\pi}} = 1.1284 \frac{Q}{k} \sqrt{\alpha t}$$

B. Heat flow to the solid bounded internally by a circular cylinder (radius = a).

1. Temperature at $r > a$ in the operational form

$$\bar{\theta} = \frac{Q}{k} \frac{K_0(qr)}{pq K_1(qa)}, \quad q^2 = P/\alpha$$

Solution of this operational form can be made for small values of time by use of asymptotic expansions of the Bessel functions and the use of a Laplace transform

2. Temperature at $r = a$.

$$\bar{\theta} = \frac{Q}{k} \frac{K_0(qa)}{pq K_1(qa)}$$

Solution for small values of time:

$$\theta = \frac{aQ}{k} \left(\frac{2\sqrt{B}}{\sqrt{\pi}} - \frac{B}{2} + \frac{B^{3/2}}{2\pi^{1/2}} - \frac{3}{16} B^2 \right)$$

where $B = \frac{\alpha t}{a^2}$ and is restricted to values

less than 0.3. Values of the function of B in the parenthesis appear in Table 1.

C. Heat flow to the solid bounded internally by a sphere (radius = a).

1. Temperature at radius $r > a$.

$$\theta = \frac{Qa}{k} \left\{ \operatorname{erfc} \left(\frac{r-a}{2\sqrt{\alpha t}} \right) - \frac{\left(\frac{r-a}{a} + \frac{\alpha t}{a^2} \right)}{2} \operatorname{erfc} \left(\frac{r-a}{2\sqrt{\alpha t}} + \sqrt{\frac{\alpha t}{a^2}} \right) \right\}$$

Values of 'erfc' of the argument are shown in tabular form in Table 2.

2. Theorem 2.1. Let $\alpha > 0$.

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

3. Let $f(x)$ be a function satisfying $f(x) = o(1)$ as $x \rightarrow \infty$.

4. Theorem 2.2. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

5. Theorem 2.3. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$. Then the function $f(x)$ is bounded as $x \rightarrow \infty$.

6. Theorem 2.4. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

7. Theorem 2.5. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

8. Theorem 2.6. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

9. Theorem 2.7. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

10. Theorem 2.8. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

11. Theorem 2.9. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

$$\sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) = \sqrt{\frac{x}{\pi}} \exp\left(-\frac{x}{2}\right) + o(1)$$

12. Theorem 2.10. Let $\alpha > 0$ and $f(x) = o(1)$ as $x \rightarrow \infty$.

2. Temperature at radius $r = a$.

$$\theta = \frac{Qa}{k} \left[1 - \frac{\frac{\alpha t}{a^2}}{e^{\frac{\alpha t}{a^2}}} \operatorname{erfc} \left(\frac{\alpha t}{a^2} \right)^{1/2} \right]$$

- II. Heat transferred to a solid bounding an elementary shape with a constant surface temperature and the initial temperature of the solid equal to zero on an arbitrary datum plane. It must be noted that for these conditions the heat flux is infinite for time equal to zero and therefore the equations are valid for t greater than zero.

- A. Linear heat flow to a medium of semi-infinite depth from a flat plane.

1. Temperature at depth x .

$$\theta = V \operatorname{erfc} \frac{x}{2 \sqrt{\alpha t}}$$

2. Heat flux at $x = 0$

$$H = \frac{kV}{\sqrt{\pi \alpha t}}$$

- B. Heat flow to a solid bounded internally by a circular cylinder (radius = a)

1. Temperature at radius $r > a$.

$$\theta = V \left(\frac{a}{r} \right)^{1/2} \operatorname{erfc} \left(\frac{r-a}{2 \sqrt{\alpha t}} \right) + \frac{V(r-a)}{8 \sqrt{\frac{a r^3}{\alpha t}}} \left(2 \operatorname{ierfc} \frac{r-a}{2 \sqrt{\alpha t}} \right)$$

reasonable for $t < 20,000$ hrs., and $a > 10$ ft.

2. Heat flux at $r = a$

$$H = \frac{4kV}{a\pi^2} \int_0^\infty \frac{e^{-\alpha u^2 t}}{J_0^2(ua) + Y_0^2(ua)} \frac{du}{u}$$

$$= \frac{4kV}{a\pi^2} I \left(0, 1; \frac{\alpha t}{a^2} \right)$$

where values of the integral

$I \left(0, 1; \frac{\alpha t}{a^2} \right)$ appear in tabular form

in Table 2.

C. Heat flow to a solid bounded internally by a sphere (radius = a).

1. Temperature at radius $r > a$.

$$\theta = \frac{aV}{r} \operatorname{erfc} \frac{r-a}{2\sqrt{\alpha t}}$$

2. Heat flux at $r = a$

$$H = kV \left[\frac{1}{\sqrt{\pi \alpha t}} + \frac{1}{a} \right]$$

Nomenclature:

- θ = temperature above arbitrary datum plane, °F
- V = temperature at $x=0$ or $r=a$ for constant temperature case, °F
- Q = constant heat flux, BTU/hr-ft²
- H = heat flux (variable), BTU/hr-ft²
- t = time, hrs
- k = thermal conductivity, BTU/hr-ft°F
- α = thermal diffusivity, ft²/hr
- x = depth into semi-infinite solid, ft.
- a = radius of circular cylinder or sphere, ft.
- r = radius of concentric cylinder or sphere composed of solid at $r > a$, ft.

$$B = \frac{\alpha t}{a^2}, \text{ dimensionless}$$

$$\operatorname{erfc} y = \frac{2}{\sqrt{\pi}} \int_y^{\infty} e^{-\beta^2} d\beta$$

$$\operatorname{ierfc} y = \frac{1}{\sqrt{\pi}} e^{-y^2} - y \operatorname{erfc} y$$

IV. REVISIONS OF ENGINEERING MANUAL

Part XVI - Chapter 3

Certain tentative revisions in the Engineering Manual, XVI Chapter 3, "Heating, Ventilating and Moisture Control", were made by B. A. Peavy of this Bureau and J. C. Letts of the O.C.E. The additions to the manual were mainly concerned with three conditions of occupancy for which the performance of an air conditioning system and related equipment must be designed. Some deletions were made in the manual because of their redundancy. Copies of the manual with these tentative revisions were presented to interested parties for their comment and review. Further work to be accomplished in writing of the manual will be concerned with work now being performed by this Bureau.

V. FUTURE EXPERIMENTATION - MT. WEATHER, VA.

1. Underground Test Chamber

Test Condition 14 - Test condition 3 repeat temperature drop with minimum heat supply and no ventilation.

Test Condition 15 - Apply refrigeration to chamber until room temperature reaches 40°F.

Test Condition 16 - Test condition 15 with use of ventilation air.

$$\begin{aligned}
 & \frac{1}{\pi} \int_0^{\pi} \frac{1}{1 - 2a \cos \theta + a^2} d\theta = \frac{1}{1 - a^2} \\
 & \frac{1}{\pi} \int_0^{\pi} \frac{\cos n\theta}{1 - 2a \cos \theta + a^2} d\theta = \frac{a^n}{1 - a^2} \\
 & \frac{1}{\pi} \int_0^{\pi} \frac{\sin n\theta}{1 - 2a \cos \theta + a^2} d\theta = 0
 \end{aligned}$$

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Test Condition 17 - Repeat test condition 1, constant heat input at about 75,000 BTU/hr until temperature of chamber is 75°F.

2. Underground Spray Pond Heat Exchanger

The spray pond was ready to operate before the finish of this period, but interference due to blasting by Bureau of Mines operations halted this test. Testing will begin May 11, with the test conditions for the various spray pond tests controlled as follows:

Test Condition 1 - A constant heat input rate to the spray water of approximately 60,000 BTU/hr will be used until the temperature of the pond water reaches 100°F.

Test Condition 2 - Maintain the temperature of the pond water at 100°F.

Test Conditions 3 and 4. Repeat test conditions 1 and 2 with use of stagnant pond instead of using sprays.

3. Tunnel Ventilation

Preparations are being made for determining the heat exchange between tunnel walls and an air stream. Tests will begin during the month of June.

VI. USE OF AN OCCUPIED UNDERGROUND SPACE FOR TEMPERATURE STUDIES.

Background:

Following a visit to an underground space near Fort Ritchie, Maryland, it was concluded that some of the data needed for correlation with present studies at Mount Weather, Va. could be obtained by making appropriate tests in the underground space there. Seasonal heating or cooling of the ventilation air by rock wall shafts will occur at this site and the amount of heat exchanged between ventilation air and the rock mass could be evaluated. Also studies of the heat transferred to the rock mass surrounding the occupied structure could be made.

Test Condition 1 - The test was conducted in the
test tank at about 75° F. The water level
was at about 1.5 ft.

1. The test was conducted in the test tank.

The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

Test Condition 2 - The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

Test Condition 3 - The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

Test Condition 4 - The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

2. The test was conducted in the test tank.

The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

3. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

Background

Following a visit to the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft. The test was conducted in the test tank at about 75° F. The water level was at about 1.5 ft.

Proposed Work to be Done:

Temperature sensing elements will be installed on the rock surface and at selected depths in the rock in the ventilation air shafts and inside and outside of the structure proper. Other instruments will be provided to measure air flow, heat flux to rock mass, and relative humidity. These instruments and their locations will be carefully selected so that measurements would provide data for computations of heat transfer rates. During this reporting period thermocouples were installed at six positions on the rock surface and at selected depths in the rock. The remaining work will be finished during the month of June.

TABLE 1

B	f (B)	B	f (B)
.001	.0321	0.050	.2302
.002	.0495	0.052	.2341
.004	.0694	0.054	.2382
.006	.0844	0.056	.2422
.008	.0971	0.058	.2461
.010	.1081	0.060	.2499
.012	.1176		
.014	.1269	0.065	.2591
.016	.1348	0.070	.2677
.018	.1430	0.075	.2765
.020	.1503	0.080	.2843
.022	.1572	0.085	.2921
.024	.1636	0.090	.2996
.026	.1700		
.028	.1760	0.10	.3138
.030	.1819	0.11	.3273
.032	.1859	0.12	.3401
.034	.1910	0.13	.3519
.036	.1963	0.14	.3633
.038	.2009	0.15	.3741
.040	.2074	0.16	.3847
.042	.2123	0.17	.3937
.044	.2210		
.048	.2258		

TABLE 1

ITEM	QTY	UNIT	PRICE	TOTAL
ITEM 1	100	EA	1.00	100.00
ITEM 2	200	EA	2.00	400.00
ITEM 3	300	EA	3.00	900.00
ITEM 4	400	EA	4.00	1600.00
ITEM 5	500	EA	5.00	2500.00
ITEM 6	600	EA	6.00	3600.00
ITEM 7	700	EA	7.00	4900.00
ITEM 8	800	EA	8.00	6400.00
ITEM 9	900	EA	9.00	8100.00
ITEM 10	1000	EA	10.00	10000.00
ITEM 11	1100	EA	11.00	12100.00
ITEM 12	1200	EA	12.00	14400.00
ITEM 13	1300	EA	13.00	16900.00
ITEM 14	1400	EA	14.00	19600.00
ITEM 15	1500	EA	15.00	22500.00
ITEM 16	1600	EA	16.00	25600.00
ITEM 17	1700	EA	17.00	28900.00
ITEM 18	1800	EA	18.00	32400.00
ITEM 19	1900	EA	19.00	36100.00
ITEM 20	2000	EA	20.00	40000.00
ITEM 21	2100	EA	21.00	44100.00
ITEM 22	2200	EA	22.00	48400.00
ITEM 23	2300	EA	23.00	52900.00
ITEM 24	2400	EA	24.00	57600.00
ITEM 25	2500	EA	25.00	62500.00
ITEM 26	2600	EA	26.00	67600.00
ITEM 27	2700	EA	27.00	72900.00
ITEM 28	2800	EA	28.00	78400.00
ITEM 29	2900	EA	29.00	84100.00
ITEM 30	3000	EA	30.00	90000.00
ITEM 31	3100	EA	31.00	96100.00
ITEM 32	3200	EA	32.00	102400.00
ITEM 33	3300	EA	33.00	108900.00
ITEM 34	3400	EA	34.00	115600.00
ITEM 35	3500	EA	35.00	122500.00
ITEM 36	3600	EA	36.00	129600.00
ITEM 37	3700	EA	37.00	136900.00
ITEM 38	3800	EA	38.00	144400.00
ITEM 39	3900	EA	39.00	152100.00
ITEM 40	4000	EA	40.00	160000.00
ITEM 41	4100	EA	41.00	168100.00
ITEM 42	4200	EA	42.00	176400.00
ITEM 43	4300	EA	43.00	184900.00
ITEM 44	4400	EA	44.00	193600.00
ITEM 45	4500	EA	45.00	202500.00
ITEM 46	4600	EA	46.00	211600.00
ITEM 47	4700	EA	47.00	220900.00
ITEM 48	4800	EA	48.00	230400.00
ITEM 49	4900	EA	49.00	240100.00
ITEM 50	5000	EA	50.00	250000.00

TABLE 2

Table of Functions Used in Heat Transfer Equations

y	erfc y	2 ierfc y	I (0, 1; y)
0.00	1.0000	1.1284	-
.01	.9887	1.1085	15.122
.02	.9774	1.0888	11.033
.03	.9662	1.0694	9.218
.04	.9549	1.0502	8.135
.05	0.9436	1.0312	7.359
.06	.9324	1.0124	6.846
.07	.9212	0.9939	6.421
.08	.9099	0.9756	6.076
.09	.8987	0.9575	5.790
0.10	.8875	0.9396	5.549
.11	.8764	0.9220	5.340
.12	.8633	0.9046	5.158
.13	.8542	0.8874	4.998
.14	.8431	0.8704	4.854
.15	.8320	0.8537	4.726
.16	.8210	0.8371	4.609
.17	.8101	0.8208	4.503
.18	.7991	.8047	4.405
.19	.7882	.7889	4.315
0.20	.7773	.7732	4.232
.21	.7665	.7578	4.155
.22	.7557	.7426	4.083
.23	.7450	.7275	4.016
.24	.7343	.7128	3.953
.25	.7237	.6982	3.894
.26	.7131	.6838	3.838
.27	.7026	.6697	3.785
.28	.6922	.6557	3.735
.29	.6818	.6420	3.688
.30	.6714	.6284	3.643
.31	.6611	.6151	3.600
.32	.6509	.6020	3.559
.33	.6408	.5891	3.520
.34	.6307	.5764	3.482
.35	.6206	.5639	3.446
.36	.6106	.5515	3.412
.37	.6008	.5394	3.379
.38	.5909	.5275	3.348
.39	.5813	.5158	3.317

Table of Position Data in the Transfer Process

1	2	3	4
0.00	0.00	0.00	0.00
0.01	0.01	0.01	0.01
0.02	0.02	0.02	0.02
0.03	0.03	0.03	0.03
0.04	0.04	0.04	0.04
0.05	0.05	0.05	0.05
0.06	0.06	0.06	0.06
0.07	0.07	0.07	0.07
0.08	0.08	0.08	0.08
0.09	0.09	0.09	0.09
0.10	0.10	0.10	0.10
0.11	0.11	0.11	0.11
0.12	0.12	0.12	0.12
0.13	0.13	0.13	0.13
0.14	0.14	0.14	0.14
0.15	0.15	0.15	0.15
0.16	0.16	0.16	0.16
0.17	0.17	0.17	0.17
0.18	0.18	0.18	0.18
0.19	0.19	0.19	0.19
0.20	0.20	0.20	0.20
0.21	0.21	0.21	0.21
0.22	0.22	0.22	0.22
0.23	0.23	0.23	0.23
0.24	0.24	0.24	0.24
0.25	0.25	0.25	0.25
0.26	0.26	0.26	0.26
0.27	0.27	0.27	0.27
0.28	0.28	0.28	0.28
0.29	0.29	0.29	0.29
0.30	0.30	0.30	0.30
0.31	0.31	0.31	0.31
0.32	0.32	0.32	0.32
0.33	0.33	0.33	0.33
0.34	0.34	0.34	0.34
0.35	0.35	0.35	0.35
0.36	0.36	0.36	0.36
0.37	0.37	0.37	0.37
0.38	0.38	0.38	0.38
0.39	0.39	0.39	0.39
0.40	0.40	0.40	0.40
0.41	0.41	0.41	0.41
0.42	0.42	0.42	0.42
0.43	0.43	0.43	0.43
0.44	0.44	0.44	0.44
0.45	0.45	0.45	0.45
0.46	0.46	0.46	0.46
0.47	0.47	0.47	0.47
0.48	0.48	0.48	0.48
0.49	0.49	0.49	0.49
0.50	0.50	0.50	0.50
0.51	0.51	0.51	0.51
0.52	0.52	0.52	0.52
0.53	0.53	0.53	0.53
0.54	0.54	0.54	0.54
0.55	0.55	0.55	0.55
0.56	0.56	0.56	0.56
0.57	0.57	0.57	0.57
0.58	0.58	0.58	0.58
0.59	0.59	0.59	0.59
0.60	0.60	0.60	0.60
0.61	0.61	0.61	0.61
0.62	0.62	0.62	0.62
0.63	0.63	0.63	0.63
0.64	0.64	0.64	0.64
0.65	0.65	0.65	0.65
0.66	0.66	0.66	0.66
0.67	0.67	0.67	0.67
0.68	0.68	0.68	0.68
0.69	0.69	0.69	0.69
0.70	0.70	0.70	0.70
0.71	0.71	0.71	0.71
0.72	0.72	0.72	0.72
0.73	0.73	0.73	0.73
0.74	0.74	0.74	0.74
0.75	0.75	0.75	0.75
0.76	0.76	0.76	0.76
0.77	0.77	0.77	0.77
0.78	0.78	0.78	0.78
0.79	0.79	0.79	0.79
0.80	0.80	0.80	0.80
0.81	0.81	0.81	0.81
0.82	0.82	0.82	0.82
0.83	0.83	0.83	0.83
0.84	0.84	0.84	0.84
0.85	0.85	0.85	0.85
0.86	0.86	0.86	0.86
0.87	0.87	0.87	0.87
0.88	0.88	0.88	0.88
0.89	0.89	0.89	0.89
0.90	0.90	0.90	0.90
0.91	0.91	0.91	0.91
0.92	0.92	0.92	0.92
0.93	0.93	0.93	0.93
0.94	0.94	0.94	0.94
0.95	0.95	0.95	0.95
0.96	0.96	0.96	0.96
0.97	0.97	0.97	0.97
0.98	0.98	0.98	0.98
0.99	0.99	0.99	0.99
1.00	1.00	1.00	1.00

TABLE 2 - continued

y	erfc y	2 ierfc y	I (0, 1; y)
.40	.5716	.5043	3.288
.41	.5620	.4929	3.259
.42	.5526	.4818	3.232
.43	.5431	.4708	3.206
.44	.5338	.4600	3.180
.45	.5245	.4495	3.156
.46	.5154	.4391	3.132
.47	.5063	.4289	3.109
.48	.4973	.4188	3.086
.49	.4884	.4090	3.065
.50	.4795	.3993	3.044
.51	.4708	.3898	3.023
.52	.4621	.3805	3.003
.53	.4535	.3713	2.984
.54	.4451	.3623	2.965
.55	.4367	.3535	2.947
.56	.4284	.3448	2.929
.57	.4202	.3364	2.912
.58	.4121	.3280	2.895
.59	.4041	.3199	2.878
.60	.3961	.3119	2.862
.61	.3883	.3040	2.847
.62	.3806	.2963	2.831
.63	.3729	.2888	2.816
.64	.3654	.2814	2.802
.65	.3580	.2742	2.787
.66	.3506	.2671	2.773
.67	.3434	.2602	2.760
.68	.3362	.2545	2.746
.69	.3292	.2467	2.733
0.70	.3222	.2402	2.720
.71	.3154	.2338	2.708
.72	.3086	.2276	2.695
.73	.3019	.2215	2.683
.74	.2953	.2155	2.671
.75	.2888	.2097	2.660
.76	.2825	.2040	2.648
.77	.2762	.1984	2.637
.78	.2699	.1929	2.626
.79	.2639	.1876	2.616

TABLE 2 - continued

y	erfc y	2 ierfc y	I (0, 1;y)
0.80	.2579	.1823	2.605
.81	.2519	.1772	2.595
.82	.2462	.1723	2.584
.83	.2405	.1674	2.574
.84	.2349	.1626	2.565
.85	.2293	.1580	2.555
.86	.2239	.1535	2.545
.87	.2186	.1490	2.536
.88	.2133	.1447	2.527
.89	.2082	.1405	2.518
0.90	.2031	.1364	2.509
0.91	.1981	.1324	2.500
0.92	.1932	.1285	2.492
0.93	.1884	.1247	2.483
0.94	.1837	.1209	2.475
0.95	.1791	.1173	2.467
0.96	.1746	.1138	2.459
0.97	.1701	.1103	2.451
0.98	.1658	.1070	2.443
0.99	.1615	.1037	2.435
1.00	.1573	.1005	2.427
1.02	.1492	.0944	
1.04	.1414	.0886	
1.06	.1339	.0831	
1.08	.1267	.0779	
1.10	.1197	.0729	2.357
1.12	.1132	.0683	
1.14	.1069	.0639	
1.16	.1009	.0597	
1.18	.0952	.0558	
1.20	.0897	.0521	2.259
1.25	.0771	.0438	
1.30	.0660	.0366	2.240
1.35	.0562	.0305	
1.40	.0477	.0253	2.191
1.50			2.147
1.60			2.106
1.70			2.069
1.80			2.036
1.90			2.004

y	I (0, 1; y)
2.0	1.975
2.5	1.856
3.0	1.767
3.5	1.697
4.0	1.639
4.5	1.591
5.0	1.550
6.0	1.483
7.0	1.429
8.0	1.386
9.0	1.349
10.0	1.317
20.0	1.138
30.0	1.052
40.0	0.997
60.	0.928
80	0.884
100	0.853

VII. RESULTS OF AN INITIAL WARM-UP
PERIOD - APRIL 23 TO MAY 15, 1952.

Object:

The object of this test was to determine the time needed to bring the temperature of an underground chamber up to a temperature within the human occupancy comfort range by the means of a constant heat input rate, and also to determine empirically the equations of heat transfer to the mass bounding the chamber and the relation of these empirical equations to the theoretical approaches listed in Part III of this report.

Description of Underground Chamber and Equipment:

The underground chamber, the dimensions of which are 100'x35'x10' high, is contained in and adjacent to an experimental mine operated by the Bureau of Mines at Mount Weather, Virginia. The chamber is approximately 215 feet below the surface of the ground and 1200 feet from the surface in a horizontal direction. The rock mass bounding the chamber consists mainly of greenstone with a scattering of epidote and quartz, and traces of various other minerals. Petrographically, the greenstone is a metamorphic basalt partially colored green by the presence of chlorite.

Measurements of the surface area of the walls, floor and ceiling were made and showed that the projected surface area was approximately 10,000 square feet. Physical determinations of greenstone rock from the excavation were made at this Bureau and the results were:

Density, ρ	= 186 lbs/ft ³
Specific heat, c	= 0.2 BTU/lb. °F
Thermal Conductivity (cores), k	= 1.45 BTU/hr.ft.°F
Thermal diffusivity, α	= 0.039 ft ² /hr

The apparent porosity of greenstone samples tested by the Bureau of Mines was 0.50 percent.

As shown in Figures 11 and 12, the walls and ceiling are painted white and the floor paved with concrete. Figure 1 shows plan and elevation views of the chamber and arrangement of mechanical equipment and air distribution ductwork. Air was forced by the circulating fan (1) into the ductwork past electric strip heaters (5) to the diffusers (6) and the air from the chamber was returned to the circulating fan through the air return filters (7) and the plenum chamber.

U.S. DEPARTMENT OF THE INTERIOR
BUREAU OF LAND MANAGEMENT
WASHINGTON, D.C. 20250

10/1/55

The object of this study is to determine the size
needed to insure the maintenance of an adequate number
of deer and to determine the number of deer which
can be supported by the range of a deer range. The
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Description of the range and its location

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Fifteen twelve-foot long thermocouple poles were placed at selected positions (Figure 2) in the rock. Thermocouples had been previously attached to these poles at intervals--one half foot intervals up to six feet and one foot intervals from six to twelve feet. Room air temperatures were measured at twenty plan positions and at each plan position at heights of 2, 30, 60, and 90 inches from the floor. The thermocouples were copper-constantan and temperatures were measured by them in conjunction with an indicating potentiometer located in an instrument shed built within the chamber.

Test Procedure:

With the initial temperature in the rock up to 12 feet in depth practically uniform at 53.5°F, a constant heat input of 17.8 kilowatt or 60,800 BTU/hr was supplied to the test chamber. At regular intervals during the test, temperatures of the rock surface, rock depth room air, wet and dry bulb were recorded as well as the electric energy input as measured by kilowatt-hour meters.

The test was arbitrarily terminated at the time when the average plane rock surface temperature reached 70°F. This time was 522 hours or 21.75 days.

Results:

Figure 3 is a plot of the average rock temperature (computed from the average of the temperatures on the fifteen poles) against depth in the rock, with time from the start of the test as a parameter. Figures 4-8 show temperature distributions at various crosssections in the rock.

Figure 9 is a log-log plot of time against average temperature rise above the initial rock temperature of 53.5°F with depth in the rock as a parameter, and also the average temperature rise of the room air with time. The average room air temperature was approximately 6°F above that of the rock surface temperature throughout the test and like the rock surface temperature rise plots as a substantially straight line on log-log paper. The heat transfer from the room air to the rock surface appears to obey Newtons Law:

$$Q = h A T \quad (1)$$

where the coefficient 'h' in this case was approximately 1.0 BTU/hr ft²°F.

425

An empirical equation of average rock surface temperature rise with time has been computed from the data by the method of averages, namely:

$$\theta = 0.69t^{1/2} \quad (2)$$

Referring to Part III of this report, equations relating temperature rise at the rock surface, with the elapsed time of heating for a constant heat flux into a mass approaching infinite thickness, are for the plane surface, and the cylindrical case:

$$\theta = \frac{2Q}{k} \sqrt{\frac{\alpha t}{\pi}} \quad (3)$$

$$\theta = \frac{2Q}{k} \left[\sqrt{\frac{\alpha t}{\pi}} - \frac{\alpha t}{4a} + \frac{(\alpha t)^{3/2}}{4a^2(\pi)^{1/2}} - \frac{3(\alpha t)^2}{32a^3} \right] \quad (4)$$

respectively (for nomenclature refer to Part III of this report), where the first term in the cylindrical case is the same as that for the plane surface.

Using the experimental surface temperature data, the constants determined at the Bureau for thermal conductivity and diffusivity, and assuming the equivalent cylindrical radius of the chamber was the average of the radii computed from a) the perimeter and b) the crosssectional area, the heat flux was determined to be 4.49 and 4.90 BTU/hr.ft² for the plane surface and cylindrical case, respectively. Using these values for determining temperatures at a depth in the rock the calculated values are compared with the actual experimental values in Figure 10.

Following is a table showing the heat input to the chamber from readings of watt hour meters compared with the heat stored in the rock computed from the mean temperature rise at various times from the start of the test. Also the depth of perceptible heat penetration is noted.

Time, Hrs.	Electric Heat Input, BTU	Heat in Rock BTU	Heat Penetration ft
49	3,010,300	2,900,000	5.5
100	5,960,000	6,700,000	7.8
170	10,256,000	11,450,000	9.0
290	17,928,000	19,020,000	11.2

to include a number of other cases
concerning the use of the word
"and" in the title of the report.

(1)

$$x = \frac{1}{2} \sqrt{\frac{1}{n}}$$

It is not possible to find a single
value for the parameter λ in the
above equation. The value of λ is
always less than 1, and it is
always greater than 0. The value of
 λ is always less than 1, and it is
always greater than 0.

(2)

$$x = \frac{1}{2} \sqrt{\frac{1}{n}}$$

$$\left[\frac{x}{1-x} - \frac{1}{2} \sqrt{\frac{1}{n}} \right] = \frac{x}{1-x} - \frac{1}{2} \sqrt{\frac{1}{n}}$$

The above equation is a simple
equation, and it is not possible
to find a single value for the
parameter λ in the above
equation.

It is not possible to find a single
value for the parameter λ in the
above equation. The value of λ is
always less than 1, and it is
always greater than 0. The value of
 λ is always less than 1, and it is
always greater than 0.

The above equation is a simple
equation, and it is not possible
to find a single value for the
parameter λ in the above
equation.

Year	1950	1951	1952
1950	1,000,000	1,000,000	1,000,000
1951	1,000,000	1,000,000	1,000,000
1952	1,000,000	1,000,000	1,000,000
1953	1,000,000	1,000,000	1,000,000
1954	1,000,000	1,000,000	1,000,000

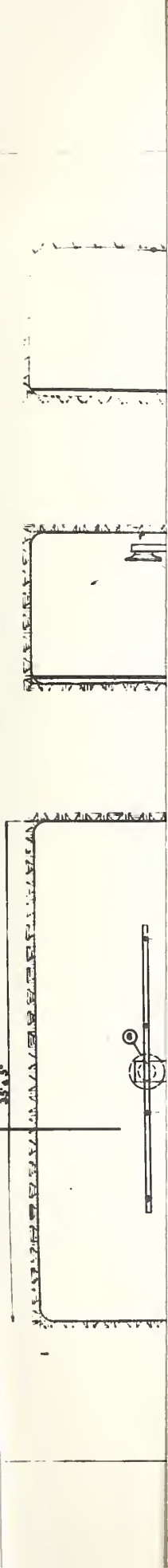
Discussion and Conclusion:

1. For the duration of this test, the temperature rise at the rock surface was proportional to the square root of the time (Equation 2). This is substantiated by theory (Equations 3 and 4) where the second, third and fourth terms of equation 4 are small in comparison to the first term for small values of time.
2. The coefficient of heat transfer between the air and rock was approximately 1.0 BTU/hr ft²(deg F). This value is approximately what would be expected for this case wherein heat transfer was by natural convection from nearly still air, with no radiation because all surfaces were nearly at the same temperature.
3. The heat balance showed that the heat stored in the rock as computed from the observed temperatures of the rock agreed to within 5% of the measured electrical heat input.
4. Figure 10 shows that the use of the equation for heat transfer to a medium surrounding a cylinder gave better agreement with the experimental data than the equation for heat transfer from a plane surface to a semi-infinite medium. (Refer to Part III of this report, sections I,A,1 and I,B,1.) For the cylindrical case, the agreement between the experimental and computed values was improved with increase in depth from the exposed rock surface.
5. Figures 4 through 8 show isotherms in the rock at various crosssections. The isotherms tend to approach elliptical shape, especially in the smaller crosssections.
6. The constant rate of energy input to the chamber, as measured electrically, was 60,800 BTU/hr, which gives an average heat flux of 6.08 BTU/hr ft² for the measured 104 square feet of projected surface area. The heat flux computed from the heat transfer equations on the basis of surface temperatures and measured rock properties was 4.49 and 4.90 BTU/hrft² for the plane surface and cylindrical equations, respectively. The discrepancy between measured and calculated heat flux is believed to be due to the fact that heat flow from the chamber surfaces took place in three dimensions (diverging as do the radii of a sphere), whereas the equations apply strictly to one-dimensional flow (plane equation) or two dimensional flow (cylinder

Discussion and Conclusions

1. The first objective of this study was to determine the effect of the concentration of the solution on the rate of the reaction. The results show that the rate of the reaction increases with the concentration of the solution. This is in agreement with the theory of the reaction, which states that the rate of the reaction is proportional to the concentration of the reactants.
2. The second objective of this study was to determine the effect of the temperature on the rate of the reaction. The results show that the rate of the reaction increases with the temperature. This is in agreement with the theory of the reaction, which states that the rate of the reaction is proportional to the temperature.
3. The third objective of this study was to determine the effect of the catalyst on the rate of the reaction. The results show that the rate of the reaction increases with the catalyst. This is in agreement with the theory of the reaction, which states that the rate of the reaction is proportional to the concentration of the catalyst.
4. The fourth objective of this study was to determine the effect of the solvent on the rate of the reaction. The results show that the rate of the reaction increases with the solvent. This is in agreement with the theory of the reaction, which states that the rate of the reaction is proportional to the concentration of the solvent.
5. The fifth objective of this study was to determine the effect of the pressure on the rate of the reaction. The results show that the rate of the reaction increases with the pressure. This is in agreement with the theory of the reaction, which states that the rate of the reaction is proportional to the pressure.

equation). The difference between the measured average input flux and that calculated from the equations is considered due to the extra bulk of rock beyond edges and corners, which caused greater flux at nearby surface areas than at other areas of the chamber, thus causing the average flux to be greater than the value at other areas where the stated equations apply more correctly. For the chamber investigated, the flux which appears appropriate for use in the cited equations was about 80% of the measured input flux.



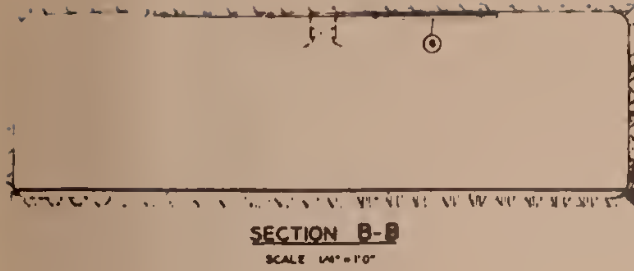
CONSTRUCTION NOTES

- A ALL SLEEVE & LINTEL OPENINGS PROVIDED BY OTHERS
- B FAN #1 SUPPORT - 2-72 LB ANGLES 4" x 3" x 7/16" 4" LEG DOWN
- C COOLING COIL SUPPORT - 2 L's 5"x1 3/4" 87 LB
- D COOLING COIL SUPPORT - 4 L's 1 1/2" x 1 1/2" x 3/10" 18 LB
- E DUCTWORK INSTALLATION TO TERMINATE AT THIS POINT INSTALLATION TO OUTSIDE BY OTHERS
- F CONDENSER WATER PIPING TO TERMINATE AT THIS POINT INSTALLATION TO COOLING TOWER BY OTHERS
- G DUCTWORK & CONDUIT SUPPORT - 8 L's 3"x3"x1/4" 20 FT LONG INSTALLATION BY OTHERS
- H DUCTWORK SUPPORT - 2 L's 3"x3"x1/4" 8 FT LONG INSTALLATION BY OTHERS

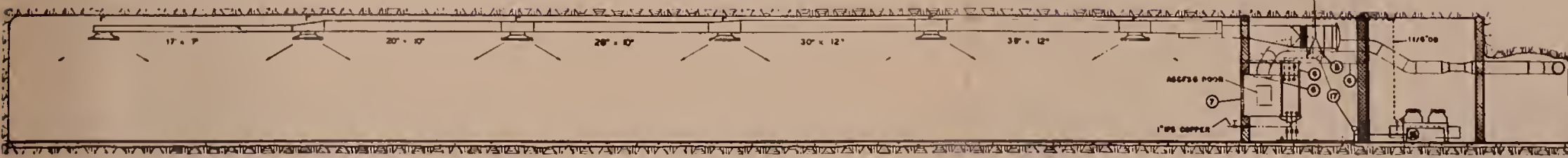
EQUIPMENT LIST

ITEM #	DESCRIPTION	QUANTITY
1	FAN AXIAL - 8000 CFM 2" TP 440 V 3 Ø 80 ~	1
2	FAN AXIAL - 800 CFM 1 1/2" TP 440 V 3 Ø 80 ~	1
3	FAN AXIAL - 800 CFM 1 1/2" TP 440 V 3 Ø 80 ~	1
4	COOLING COIL 10 Ø SEE DETAIL SHEET M-8	1
5	STRIP HEATER 440 V 3 Ø 80 ~ SEE DETAIL SHEET M-8	1
6	DIFFUSERS, 1000 CFM EACH	6
7	AIR FILTERS 5500 CFM (20" X 25")	4
8	AIR FILTERS 300 CFM (20" X 20")	2
9	10 HP SELF CONTAINED WATER CHILLER	2
10	VAREO TURBS	2
11	COOLING TOWER PUMP	1
12	STEP CONTROLLER	1
13	HUMIDISTAT	1
14	MODULATING THERMOSTAT (15-90°F)	1
15	MODULATING THERMOSTAT DUCT TYPE (15-85°F)	2
16	CHILLED WATER CONTROL (10-70°F)	1
17	THERMOMETERS (SEE DRAWING M-2)	6

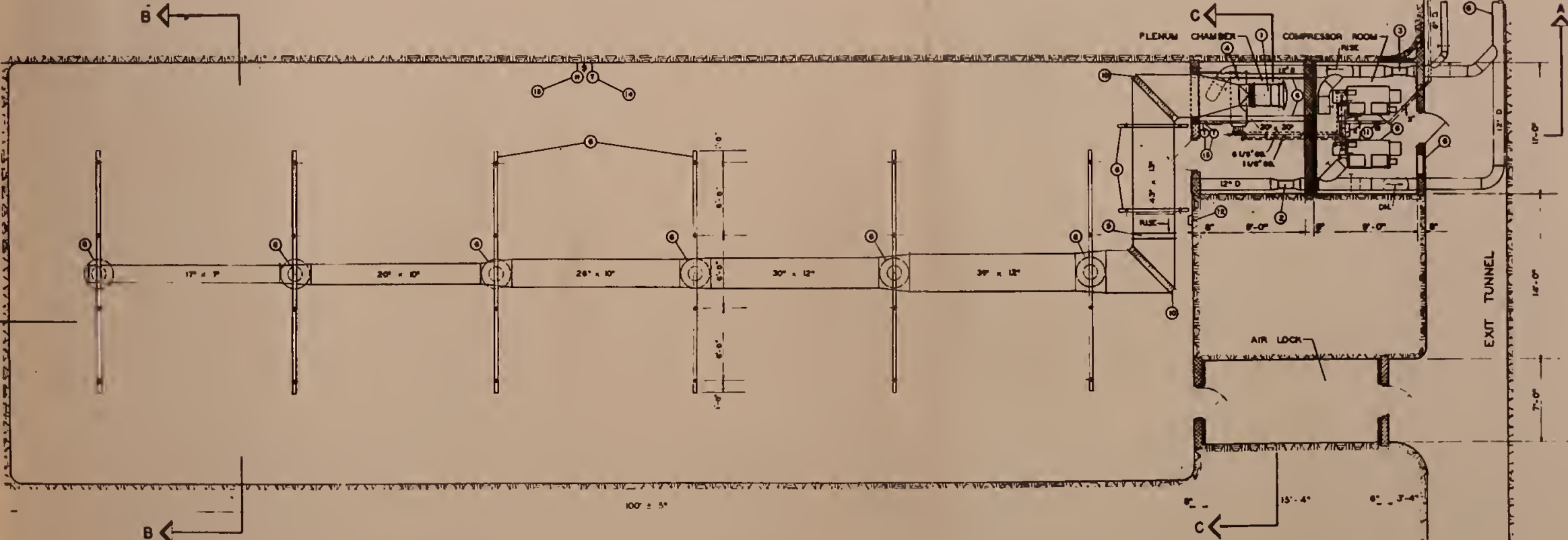
REVISION	DATE	APP'D	DESCRIPTION	BY
MECHANICAL PLAN				
NATIONAL BUREAU OF STANDARDS WASHINGTON, D.C.				
UNDERGROUND STRUCTURE PILOT PROJECT FOR THE OFFICE OF THE CHIEF OF ENGINEERS U.S. ARMY				
DESIGNED BY TERNER		DATE MAY 5 1954		
DRAWN BY S.B.		5 55 M 1		
TRACED BY S.B.		S.A. 3 0		
CHECKED BY J.L.				
APPROVED				
THOMAS H. URAHL CONTRACTOR, ENGINEER 34 JACKSON PLACE, N.E. WASHINGTON, D.C.				



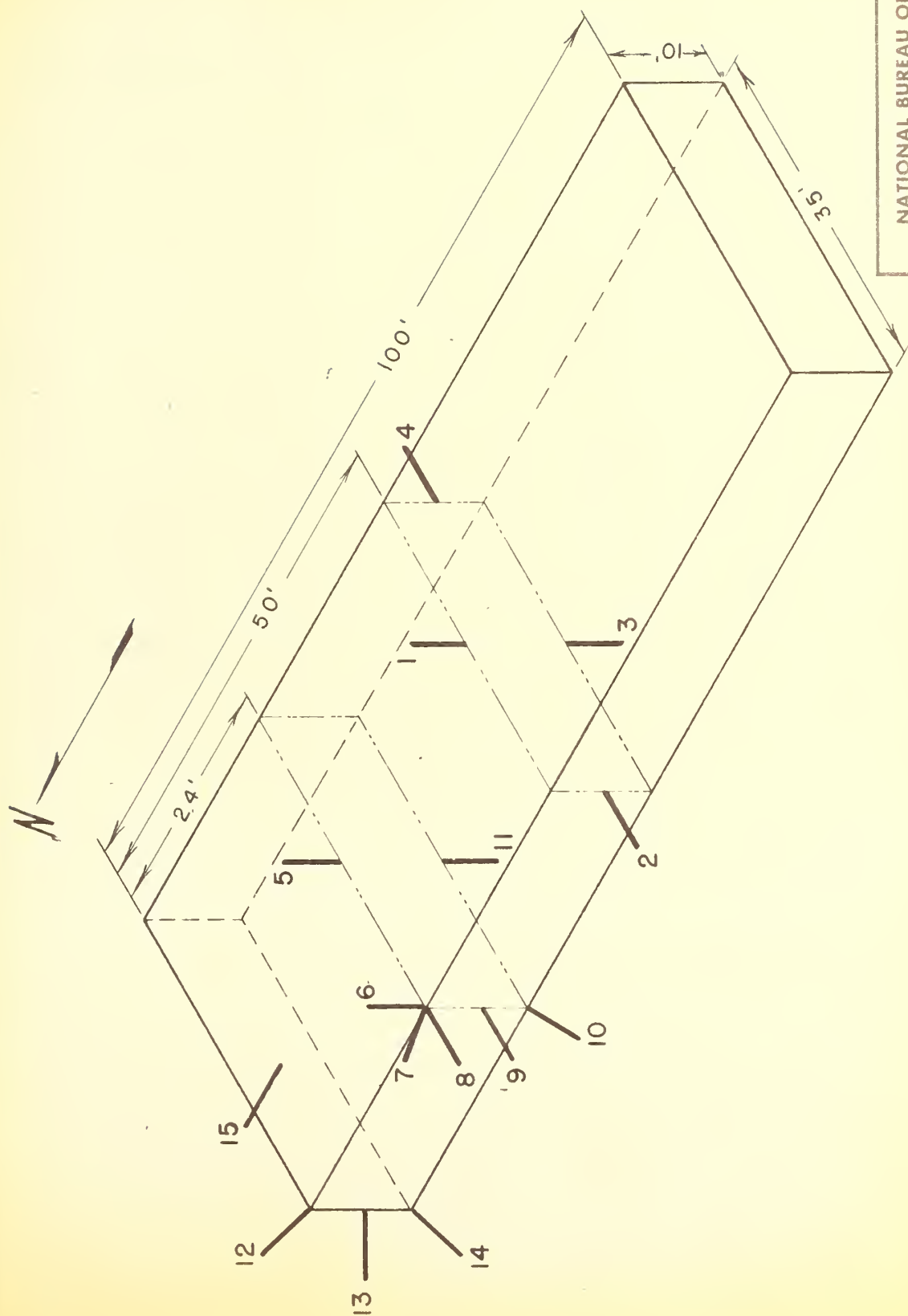
SECTION C-C
SCALE 1/4" = 1'-0"



SECTION A-A
SCALE 1/4" = 1'-0"



PLAN FIG. 1
SCALE 1/4" = 1'-0"



NATIONAL BUREAU OF STANDARDS
WASHINGTON 25, D. C.

DRAFTSMAN

DATE

SCALE

W. C. G.

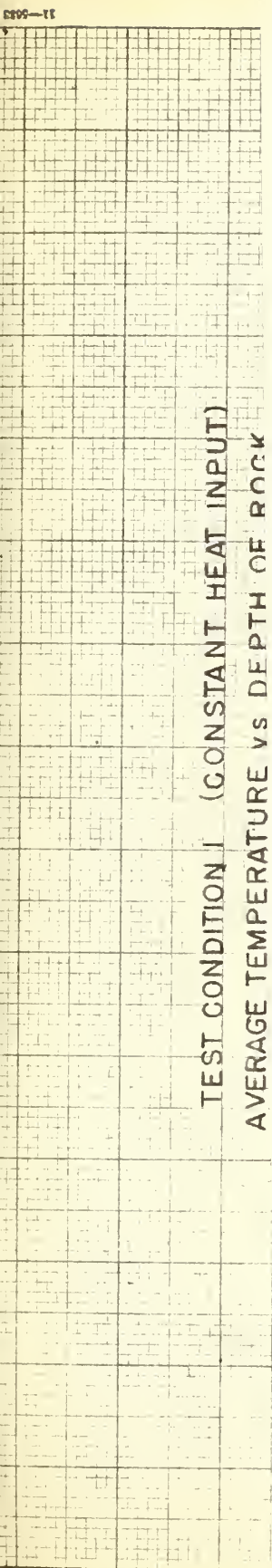
DIV SEC

UNDERGROUND
LABORATORY

FIGURE 2

70

TEST CONDITION I (CONSTANT HEAT INPUT)
AVERAGE TEMPERATURE vs DEPTH OF ROCK



TEST CONDITION 1 (CONSTANT HEAT INPUT)
AVERAGE TEMPERATURE vs DEPTH OF ROCK

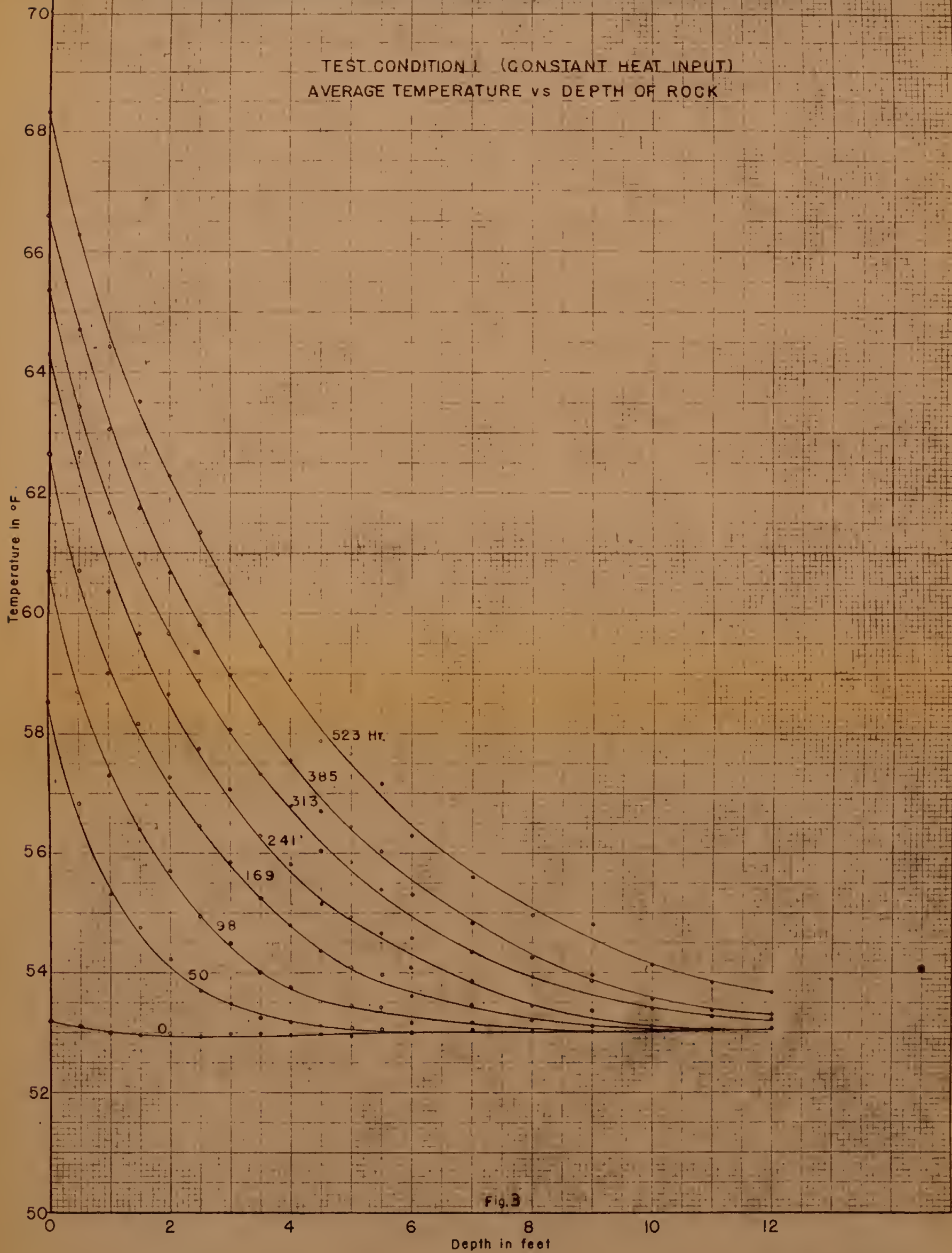
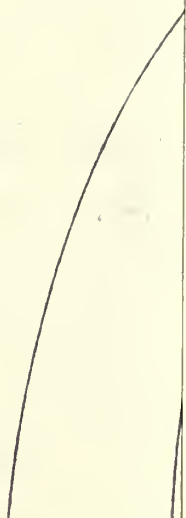


Fig. 3



2





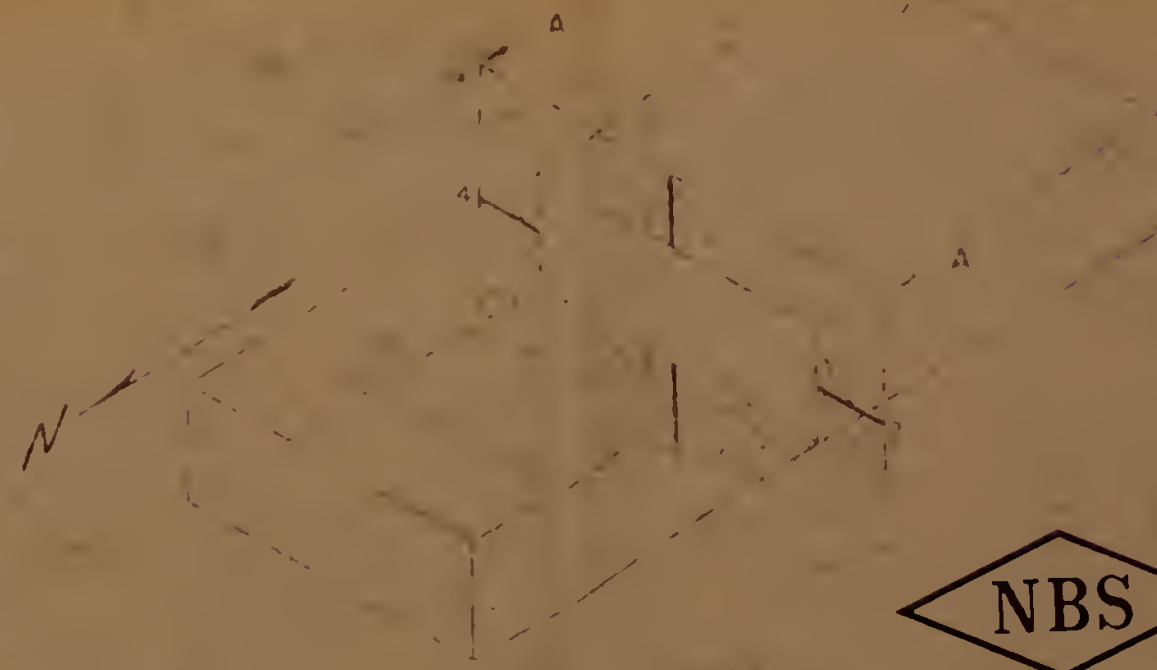


SECTION A-A

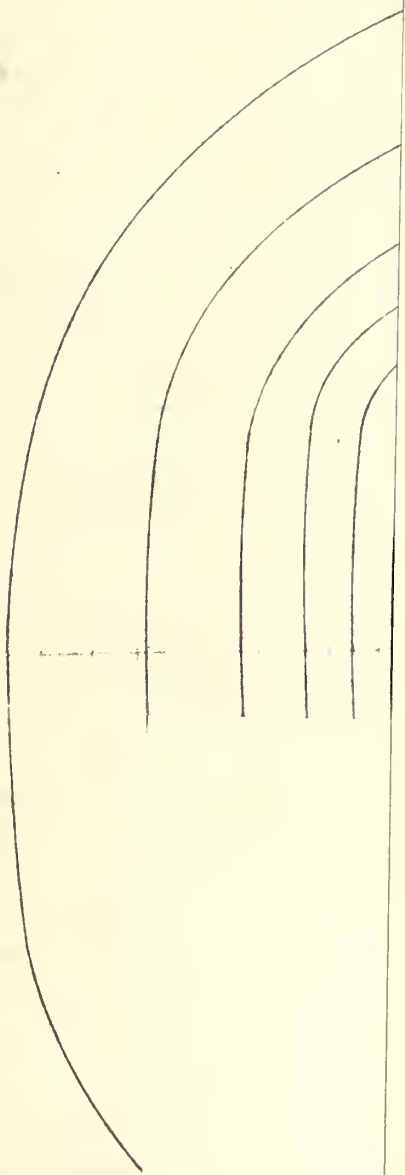
SCALE $\frac{1}{4}'' = 1'-0''$

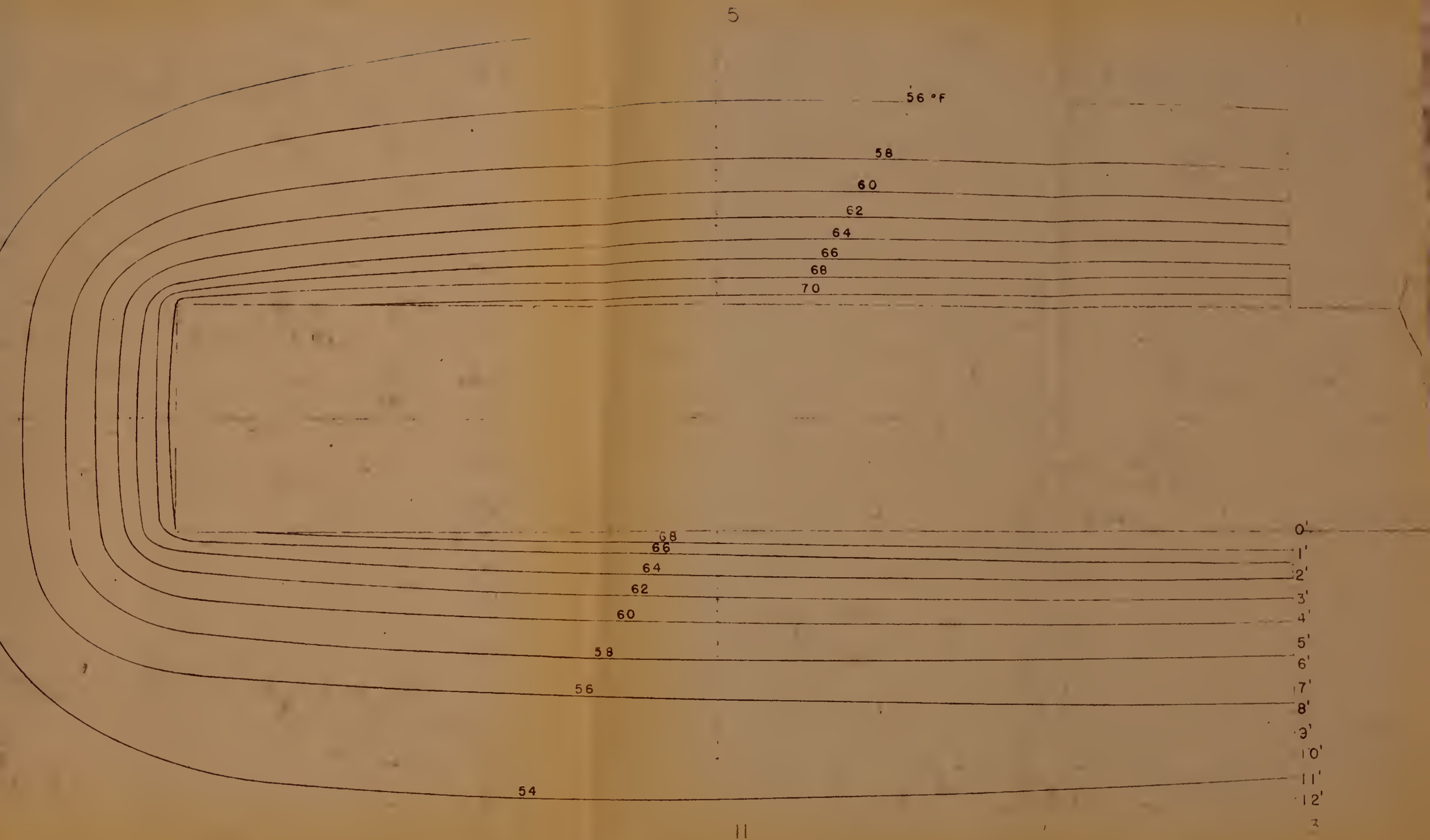
TEST CONDITION No 1- HEATING UP PHASE
TEMPERATURE DISTRIBUTION IN ROCK
AFTER 521 HOURS

FIGURE 4



NBS





SECTION 3-B

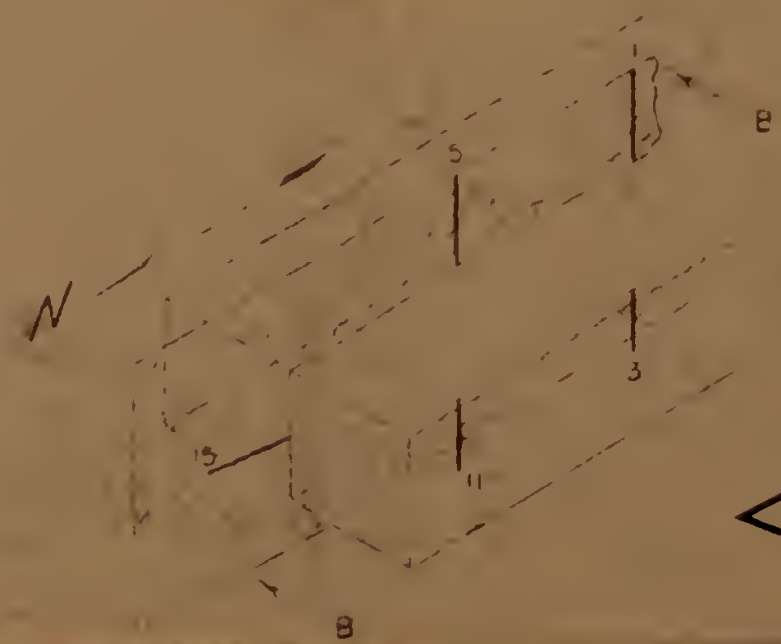
SCALE: $\frac{1}{4}" = 1'-0"$

TEST CONDITION No. 1 - HEATING UP PHASE

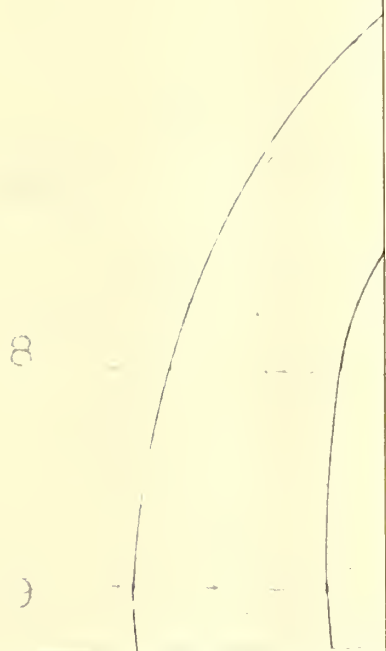
TEMPERATURE DISTRIBUTION IN ROCK

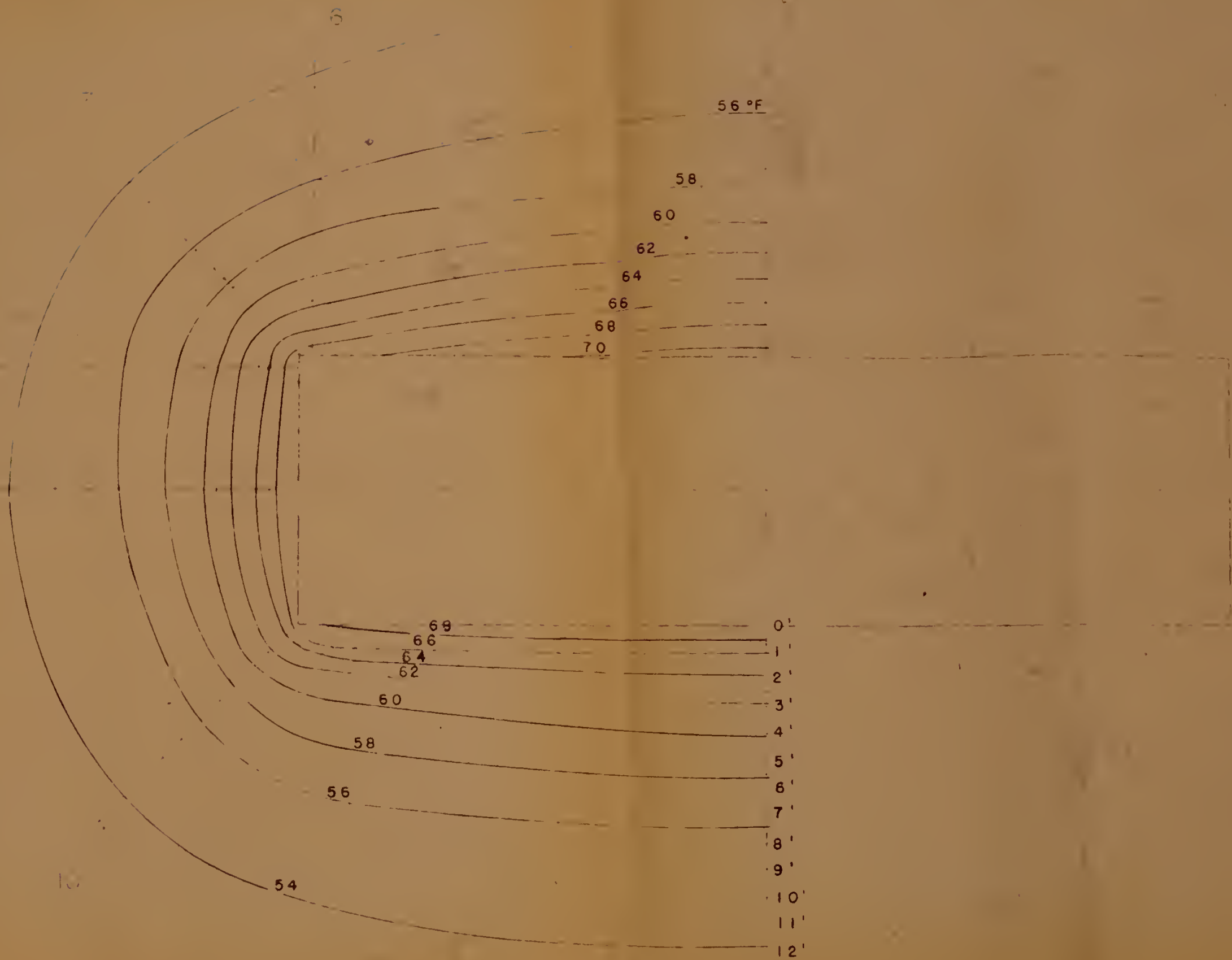
AFTER 521 HOURS

FIGURE 5



NBS





SECTION G-C

SCALE 1/4" = 1'0"

TEST CONDITION No. 1 - HEATING UP PHASE

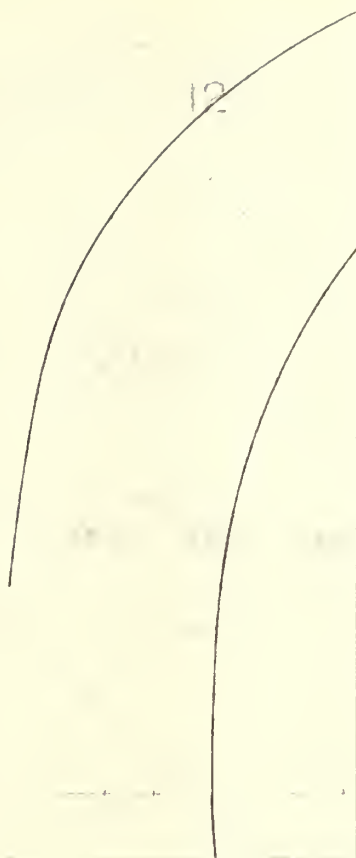
TEMPERATURE DISTRIBUTION IN ROCK

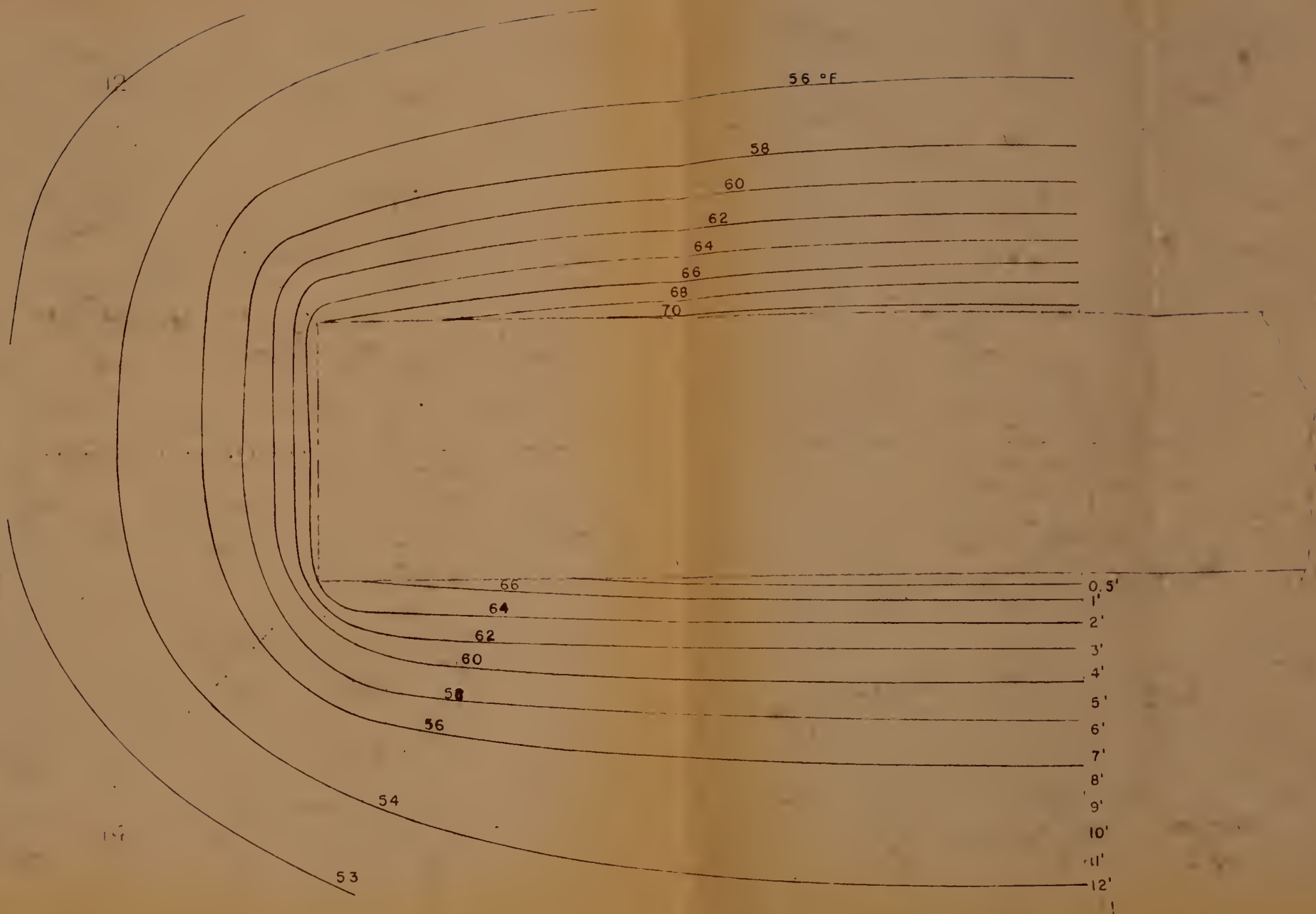
AFTER 521 HOURS

FIGURE 6



NBS



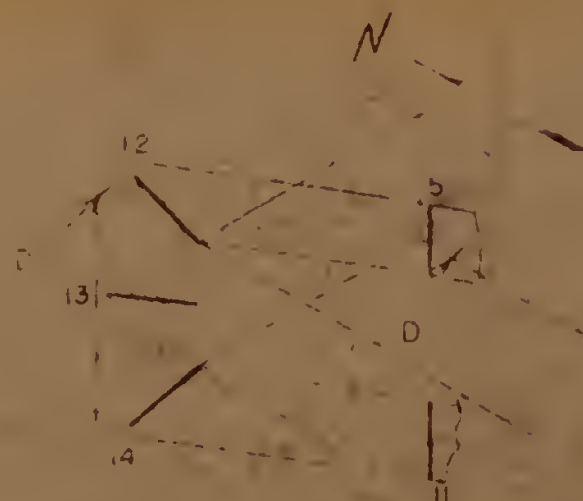


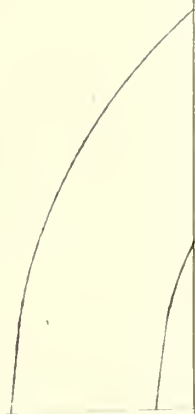
SECTION D-D

SCALE $\frac{1}{4}" = 1'-0"$

TEST CONDITION No. 1-HEATING UP PHASE
TEMPERATURE DISTRIBUTION IN ROCK
AFTER 521 HOURS

FIGURE 7





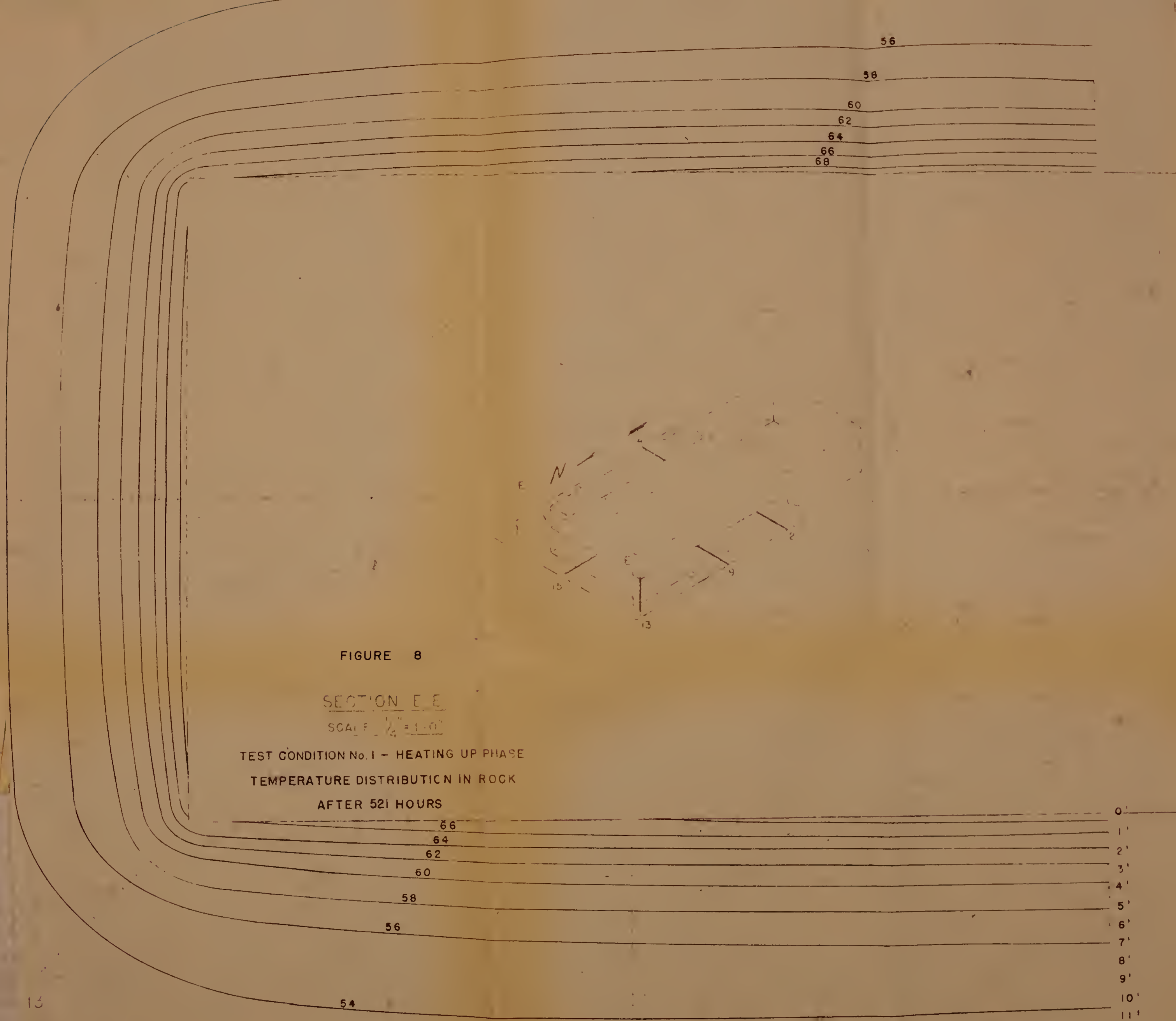


FIGURE 8

SECTION E E

SCALE $\frac{1}{4}'' = 1'-0''$

TEST CONDITION No. 1 - HEATING UP PHASE
TEMPERATURE DISTRIBUTION IN ROCK
AFTER 521 HOURS

0'
1'
2'
3'
4'
5'
6'
7'
8'
9'
10'
11'
12'
2





TIME. HOURS

FIGURE 10
T₂ FLIGHT



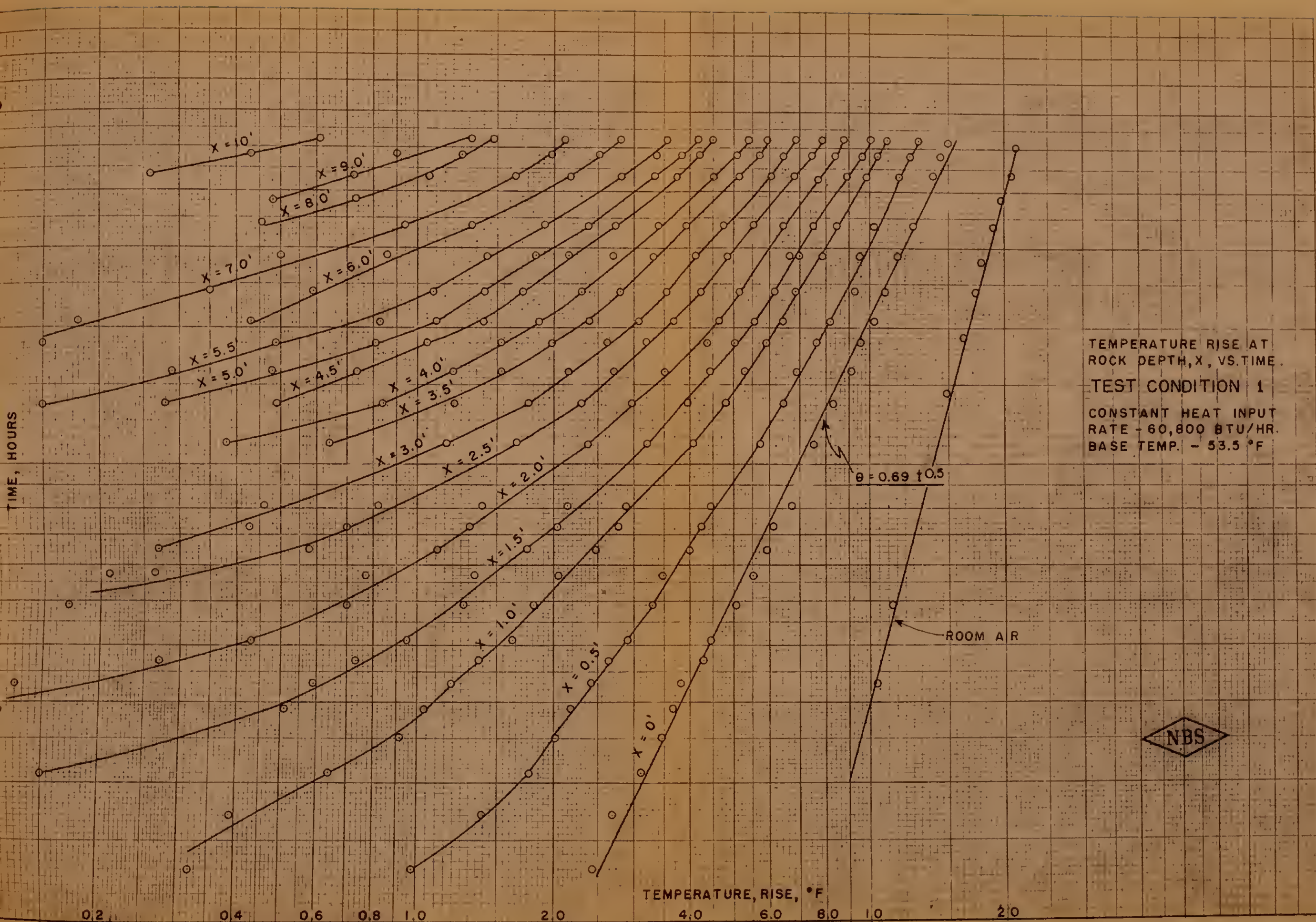
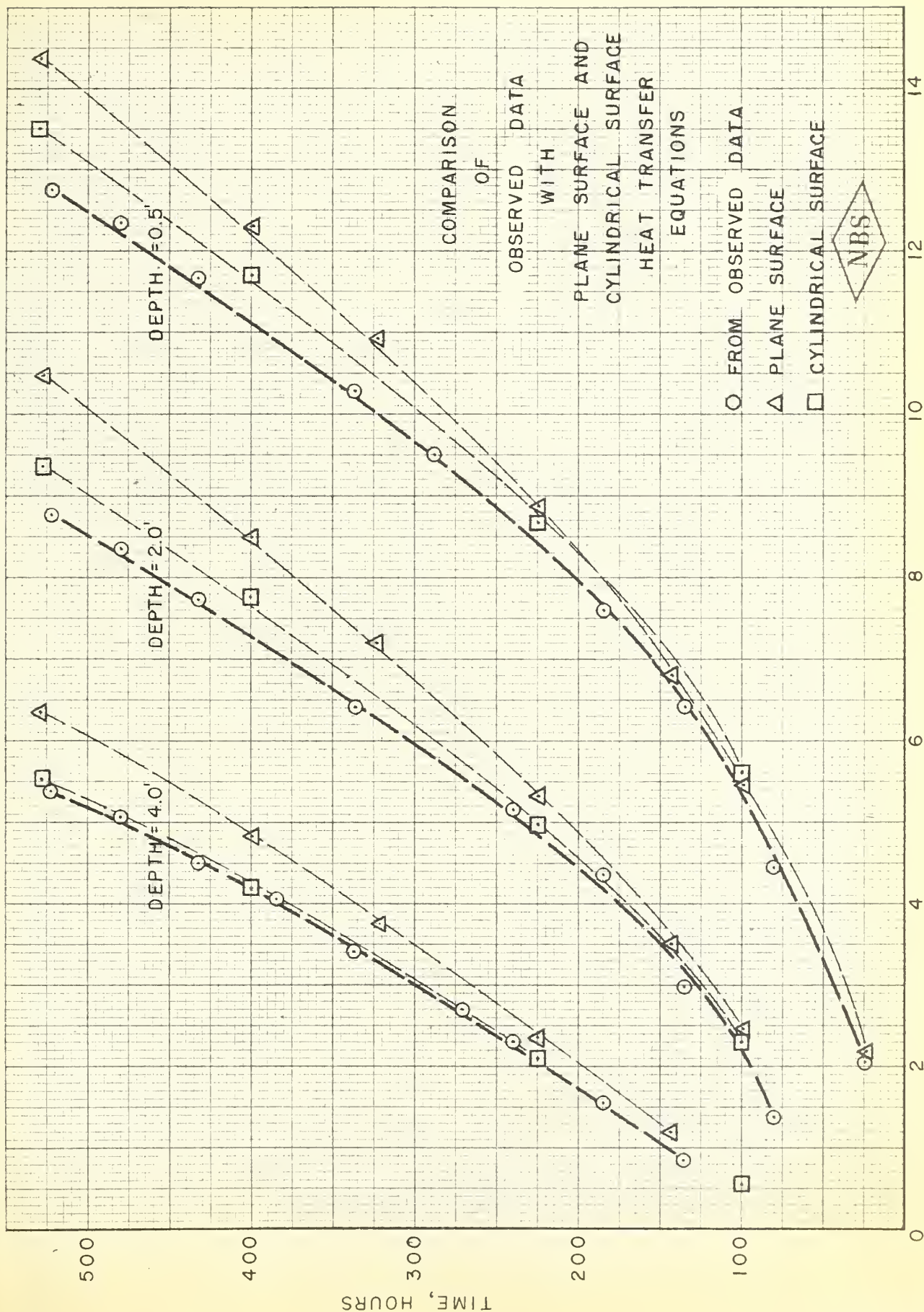


FIG. 9



TEMPERATURE DIFFERENCE $-(T_x - 53.5), ^\circ\text{F}$

FIGURE 10



Figure 11. Underground Chamber - Entrance door on right, door to plenum chamber on left.



Figure 12. Underground Chamber - Room

